Flow around Clark-Y airfoil: Summary of EFD benchmark data and comparison with IIHR EFD data and CFD solution

Nobuaki Sakamoto, J. Colin Johnson, Stuart Breczinski, Marian V. Muste and Frederick Stern IIHR-Hydroscience&Engineering, The University of Iowa

1. EFD benchmark data

[1] JACOBS E.N., STACK J., AND PINKERTON R.M., 1930 Airfoil pressure distribution investigation in the variable density wind tunnel, Langley Memorial Aeronautical Laboratory Report No. 353

[2] MARCHMAN J.F. AND WERME T.D., 1984 Clark-Y airfoil performance at low Reynolds numbers, In: proc. AIAA 22nd Aerospace Science Meeting, Jan. 9-12, Reno, Nevada, U.S.A.

[3] SILVERSTEN A., 1934, Scale effect on Clark-Y airfoil characteristics from NACA full-scale wind-tunnel tests, Langley Memorial Aeronautical Laboratory Report No. 502

[4] ZIMMERMAN C.H., Characteristics of Clark-Y airfoils of small aspect ratios, 1932, Langley Memorial Aeronautical Laboratory Report No. 431

The summary of EFD benchmark data is given in Table 1.

Reference	[1]	[2]		[3]	[4]
Digitized	C_p and C_L	C_p , C_L and C_D		C_L and C_D	C_L and C_D
data					
AR [*]	7.2	5.75		6	0.5, 0.75, 1, 1.25, 1.5, 2, 3, 6
Re ^{**}	3.56e5	C _P	7.5e4	1.12e6, 1.55e6, 2.06e6, 2.81e6,	8.6e5
		C _L , C _D	5e4, 7.5e4, 1e5, 2e5, 6.7e6	3.19e6, 3.59e6	
α (deg) ^{***}	1, 4, 7, 10,13, 17, 20	C _P C _L , C _D	0, 4, 6, 8, 12, 14 0, 4, 6, 8, 12, 14	0, 1, 2, 3, 4, 5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23	0, 10, 15, 20, 25, 30, 35, 39, 40, 42, 50, 60
Wingtip		End plate	es	Wing cross section	Rectangular

Table 1 Summary of EFD benchmark data

*: Aspect ratio, **: Reynolds number, ***: Angle of attack

2. Trend of each data set

2.1 Reference [1]

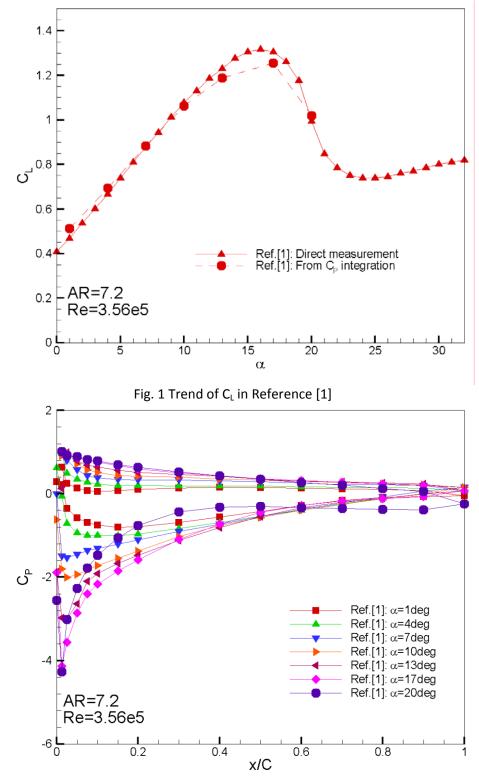


Fig. 2 Trend of C_P in Reference [1]

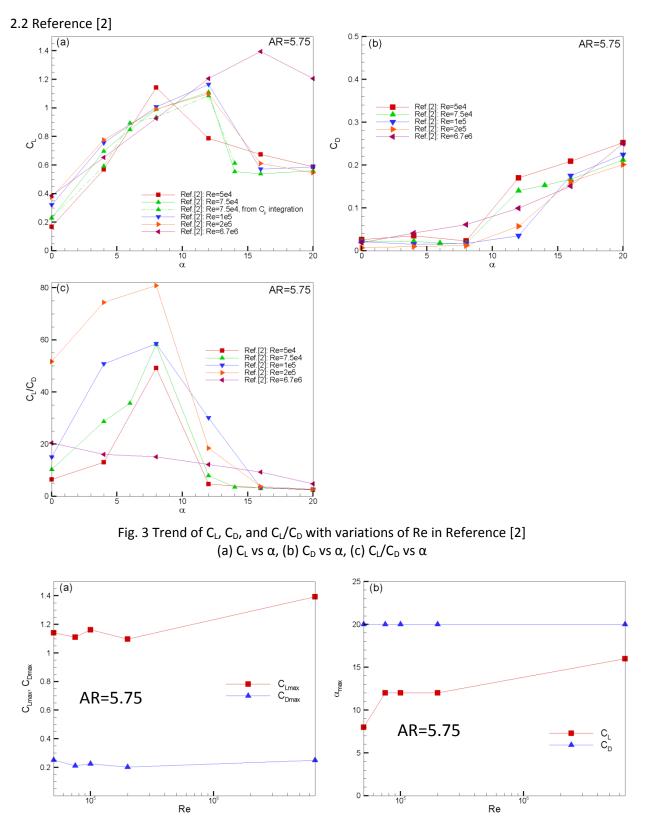


Fig. 4 Re dependency of C_{Lmax} , C_{Dmax} and α_{max} in Reference [2]: (a) C_{Lmax} , C_{Dmax} vs Re, (b) α_{max} vs Re

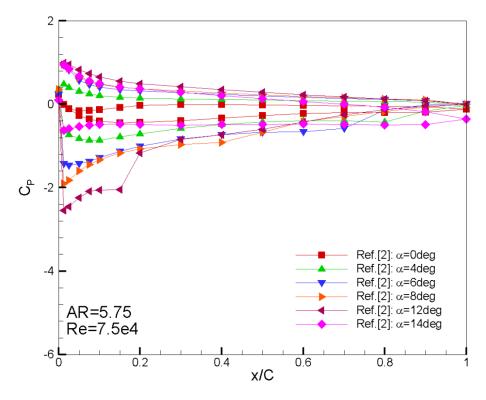


Fig. 5 Trend of C_P in Reference [2]



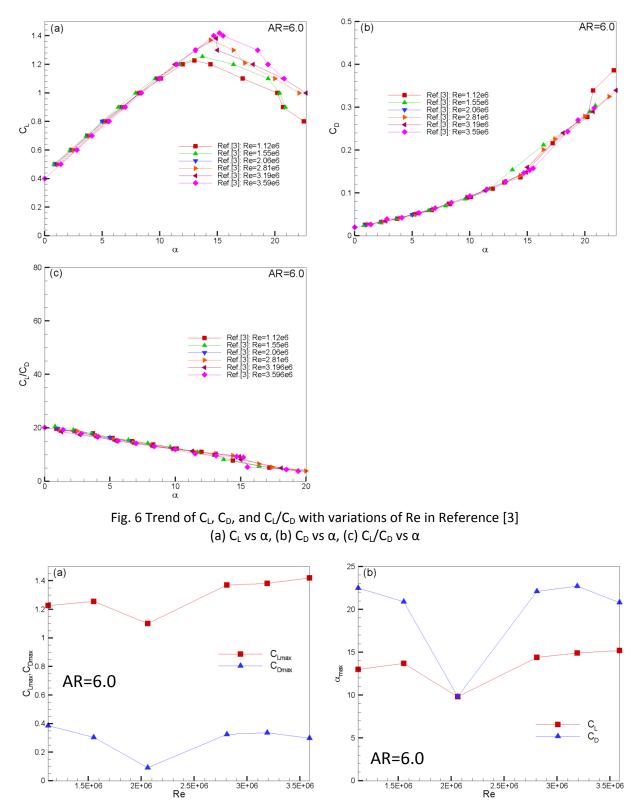
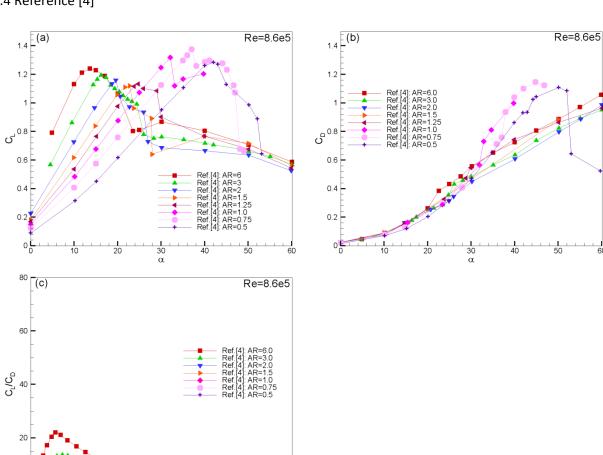
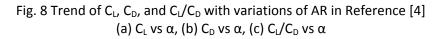


Fig. 7 Re dependency of C_{Lmax} and α_{max} for C_{L} in Reference [3]: (a) C_{Lmax} vs Re, (b) α_{max} vs Re



2.4 Reference [4]

α



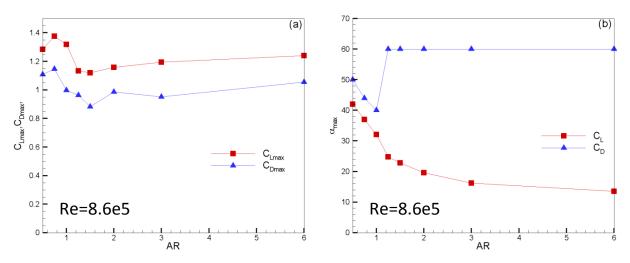
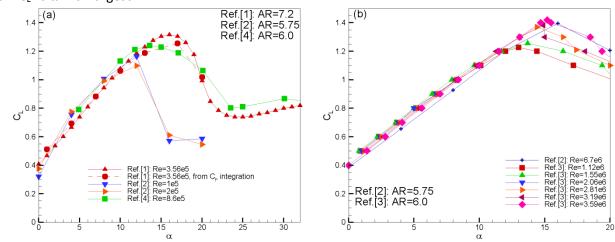
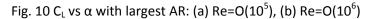


Fig. 9 Aspect ratio dependency of C_{Lmax} and α_{max} for C_{L} in Reference [4]: (a) C_{Lmax} vs AR, (b) α_{max} vs AR

3. Comparison between the reference experimental data



3.1 C_L vs α with largest AR



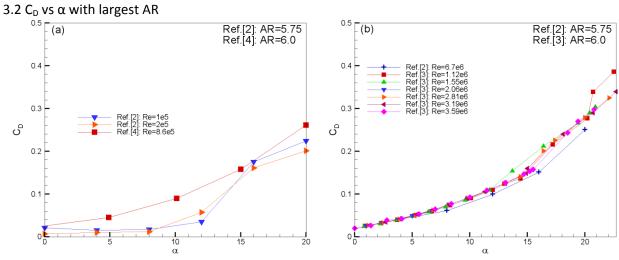


Fig. 11 C_D vs α with largest AR: (a) Re=O(10⁵), (b) Re=O(10⁶)

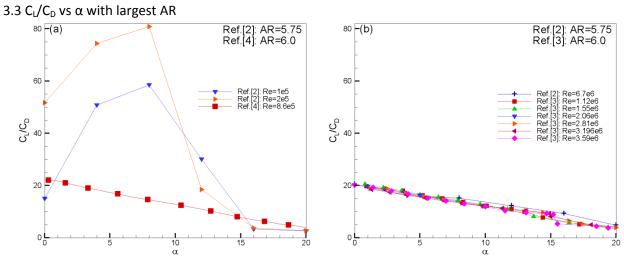


Fig. 12 C_D vs α with largest AR: (a) Re=O(10⁵), (b) Re=O(10⁶)

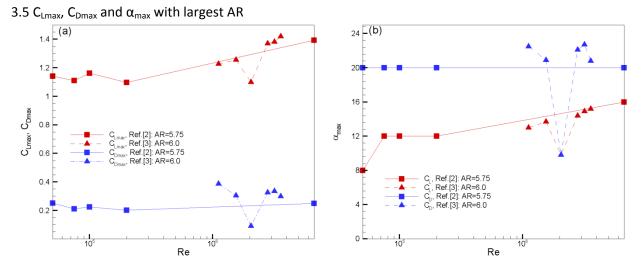


Fig. 13 Re dependency for C_{Lmax} , C_{Dmax} and α_{max} : (a) C_{Lmax} , C_{Dmax} vs Re, (b) α_{max} vs Re

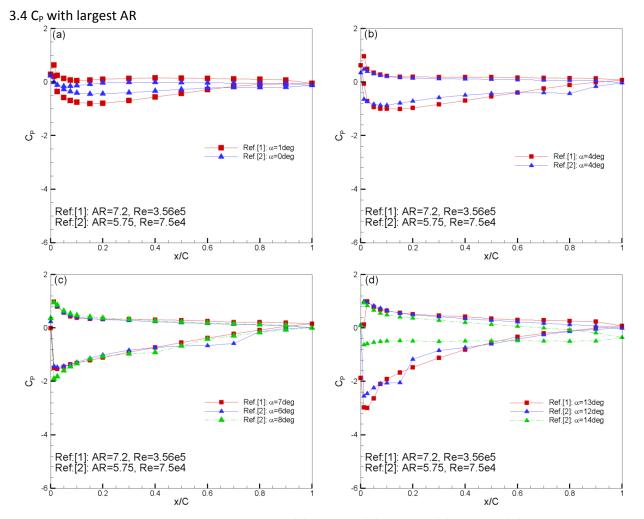
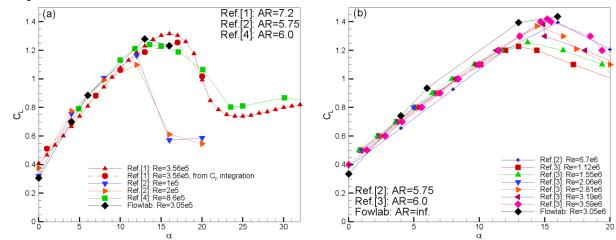


Fig. 14 C_P distribution with largest AR: (a) $\alpha \approx 0$ deg, (b) $\alpha \approx 4$ deg, (c) $\alpha \approx 7$ deg, (d) $\alpha \approx 13$ deg



4. Comparison with Flowlab simulation results 4.1 $C_L\,vs\,\alpha$

Fig. 15 C_L vs α with largest AR, Flowlab solutions added: (a) Re=O(10⁵), (b) Re=O(10⁶)

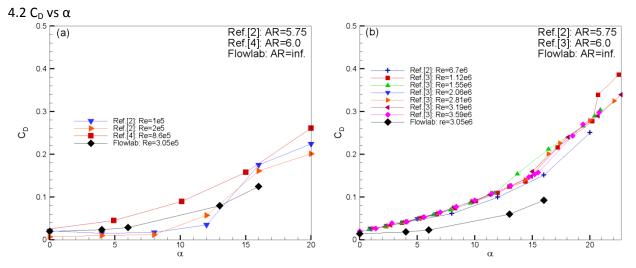


Fig. 16 C_D vs α with largest AR, Flowlab solutions added: (a) Re=O(10⁵), (b) Re=O(10⁶)

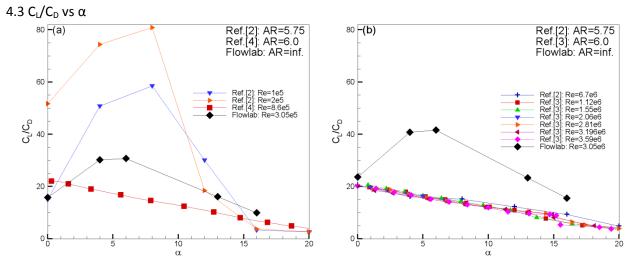
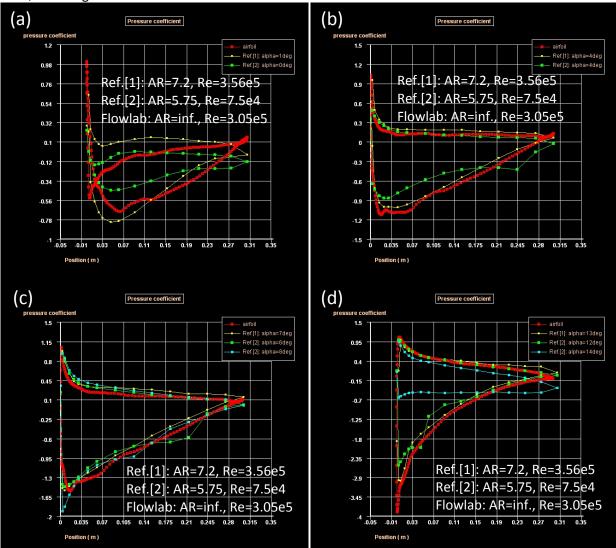
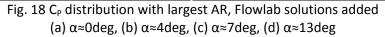
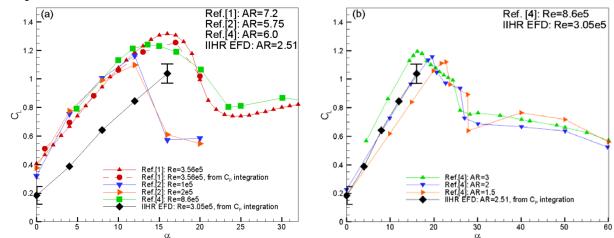


Fig. 17 C_D vs α with largest AR, Flowlab solutions added: (a) Re=O(10⁵), (b) Re=O(10⁶)



4.3 C_P with largest AR





5. Comparison between EFD benchmark data and IIHR experimental data 5.1 C_L vs α

Fig. 19 C_L vs α , IIHR EFD data added: (a) large AR (\geq 5.75), (b) small AR (1.5 \leq AR \leq 3)

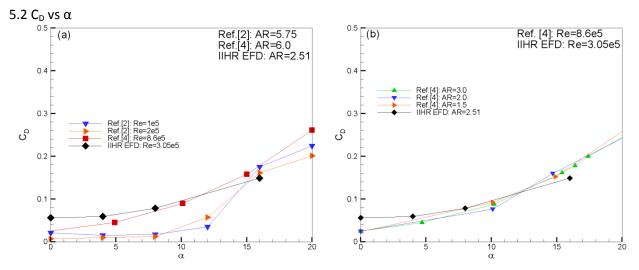


Fig. 20 C_D vs α , IIHR EFD data added: (a) large AR (\geq 5.75), (b) small AR (1.5 \leq AR \leq 3)

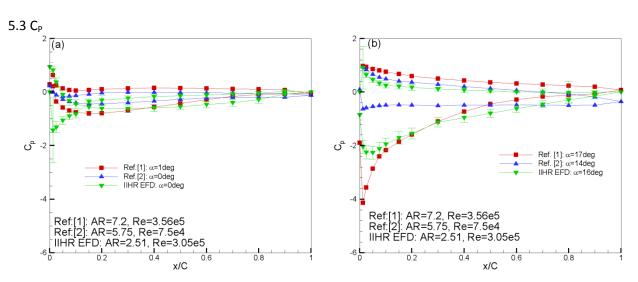
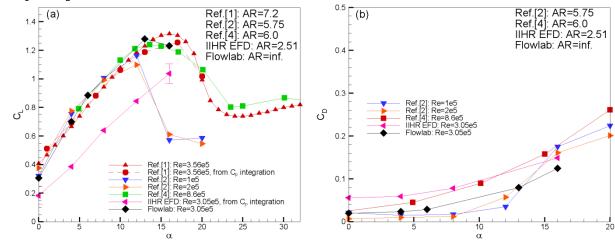


Fig. 21 C_P distribution, IIHR EFD data added: (a) $\alpha \approx 0$ deg, (b) $\alpha \approx 16$ deg



6. Comparison between EFD benchmark data, IIHR experimental data and Flowlab solution 6.1 CL and CD vs α

Fig. 22 C_L and C_D vs α : (a) C_L , (b) C_D

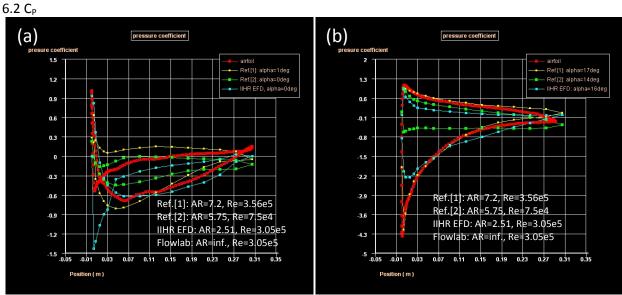


Fig. 23 C_P distribution: (a) $\alpha \approx 0$ deg, (b) $\alpha \approx 16$ deg

7. Discussion and Conclusion (To be added.)