

Non-inertial Frame of Reference

NS equations: inertial reference frame is stationary or moving at constant speed relative to stationary reference frame; for our purposes stationary reference frame = stationary with respect to distant stars

$$\nabla \cdot \underline{u} = 0 \quad \rho \frac{\partial \underline{u}}{\partial t} = -\nabla P + \rho \underline{g} + \mu \nabla^2 \underline{u} \quad \text{incompressible flow}$$

$$\text{Note: } \mu \frac{\partial^2 u_i}{\partial x_c^2} = 2\mu \frac{\partial s_{ci}}{\partial x_c} = \mu \frac{\partial}{\partial x_c} (u_{ci} + u_{ic}) = -\rho \epsilon_{ijk} \frac{\partial \omega_k}{\partial x_c}$$

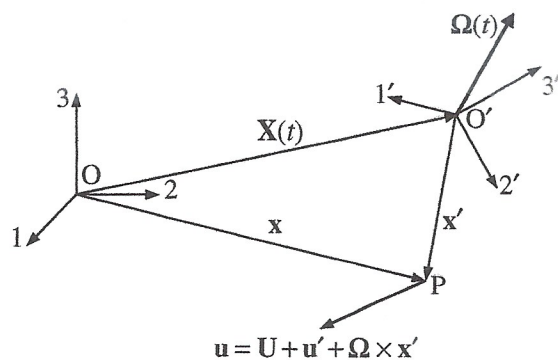
or $\mu \nabla^2 \underline{u} = -\mu \nabla \times \underline{\omega}$ (1) Seemingly paradoxical net viscous force $f(\underline{\omega})$

(2) $\mu \nabla^2 \underline{u} = 0$ when $\underline{\omega} = \text{constant}$ (solid body rotation), which also requires $\frac{\partial s_{ci}}{\partial x_c} = 0$. even though postulate #2 for $T_{ci} = f(\epsilon_{cij})$ state of rigid body rotation causes no shear stress, is resolved since involving derivatives of ω_k

However, many applications require non-inertial reference frames: rotating machinery; maneuvering vehicles; geophysical flows (atmospheric, oceanic); etc.

The continuity equation is not altered, but the NS is.

FIGURE 4.6 Geometry showing the relationship between a stationary coordinate system $O123$ and a noninertial coordinate system $O'1'2'3'$ that is moving, accelerating, and rotating with respect to $O123$. In particular, the vector connecting O and O' is $X(t)$ and the rotational velocity of $O'1'2'3'$ is $\Omega(t)$. The vector velocity u at point P in $O123$ is shown. The vector velocity u' at point P in $O'1'2'3'$ differs from u because of the motion of $O'1'2'3'$.



non-inertial

reference frame $O'1'2'3'$: + translates at $\frac{dX(t)}{dt} = U(t)$

and rotates at $\Omega(t)$ with respect

to stationary reference frame $O123$

U and Ω can be resolved in either frame however time is invariant between both reference frames, i.e., $t' = t$.

Fluid particle $P = P(x')$ or $P(x)$ where
 $x' = (x'_1, x'_2, x'_3)$ and $x = (x_1, x_2, x_3)$ and
 $x = x + x'$.

The velocity u of P is

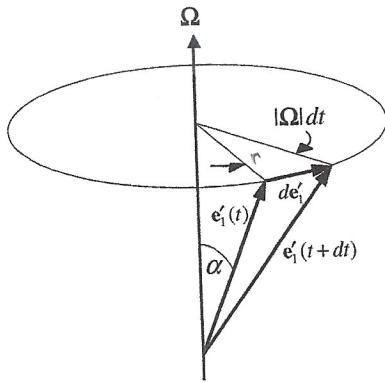
$$u = \frac{dx}{dt} = \frac{dX}{dt} + \frac{dx'}{dt} = U + \frac{d}{dt}(x'_1 e'_1 + x'_2 e'_2 + x'_3 e'_3)$$

$$= U + \frac{dx'_1}{dt} e'_1 + \frac{dx'_2}{dt} e'_2 + \frac{dx'_3}{dt} e'_3 + x'_1 \frac{de'_1}{dt} + x'_2 \frac{de'_2}{dt} + x'_3 \frac{de'_3}{dt} = U + u' + \Omega \times x'$$

u'

$\Omega \times x'$

The cross product $\Omega \times x' = x'_1 e'_{1z} + x'_2 e'_{2z} + x'_3 e'_{3z}$
 derivation is based on geometric considerations.



$$\sin \alpha = \frac{r}{|e_i|} = r$$

$$r \Omega dt = d e_i =$$

arc length
($r\theta = s$)

FIGURE 4.7 Geometry showing the relationship between Ω , the rotational velocity vector of $O'1'2'3'$, and the first coordinate unit vector e'_1 in $O'1'2'3'$. Here, the increment $d e'_1$ is perpendicular to Ω and e'_1 .

Rotation $O'1'2'3'$ per time increment dt causes, e.g., \hat{e}'_1 to trace a small portion of a cone with vertex $\sin \alpha$.

$$\underline{r} \times \underline{\Omega} = |\underline{r}| |\underline{\Omega}| \sin \alpha$$

$$d|\hat{e}'_1| = \sin \alpha |\underline{\Omega}| dt \Rightarrow \frac{d|\hat{e}'_1|}{dt} = \sin \alpha |\underline{\Omega}|$$

$$= \underline{\Omega} \times \hat{e}'_1 \quad |\hat{e}'_1| = 1$$

$$\text{ie } \frac{d\hat{e}'_i}{dt} = \underline{\Omega} \times \hat{e}'_i$$

$$\text{so } x'_1 \frac{d\hat{e}'_1}{dt} + x'_2 \frac{d\hat{e}'_2}{dt} + x'_3 \frac{d\hat{e}'_3}{dt} = \underline{\Omega} \times \underline{x}'$$

Acceleration:

$$\underline{a} = \frac{d\underline{u}}{dt} = \frac{d}{dt}(\underline{U} + \underline{u}' + \underline{\Omega} \times \underline{x}') = \frac{d\underline{U}}{dt} + \underline{a}' + 2\underline{\Omega} \times \underline{u}' + \frac{d\underline{\Omega}}{dt} \times \underline{x}' + \underline{\Omega} \times (\underline{\Omega} \times \underline{x}')$$

\underline{U}_T = acceleration O' wrt O

\underline{a}' = acceleration in O' reference frame

$2\underline{\Omega} \times \underline{u}'$ = Coriolis acceleration

$\frac{d\underline{\Omega}}{dt} \times \underline{x}'$ = acceleration in O' due $\underline{\Omega}$

$\underline{\Omega} \times (\underline{\Omega} \times \underline{x}')$ = centripetal acceleration

For fluid: $\underline{a} = \frac{d\underline{u}}{dt} = \frac{D\underline{u}}{Dt}$ & $\underline{a}' = \frac{D\underline{u}'}{Dt}$ i.e. derivatives follow the motion of the fluid particle

$$\left(\frac{D\underline{u}}{Dt}\right)_{O123} = \left(\frac{D\underline{u}'}{Dt}\right)_{O1'2'3'} + \frac{d\underline{U}}{dt} + 2\underline{\Omega} \times \underline{u}' + \frac{d\underline{\Omega}}{dt} \times \underline{x}' + \underline{\Omega} \times (\underline{\Omega} \times \underline{x}')$$

Substitute into the NS equation:

$$\rho \left(\frac{D\underline{u}'}{Dt}\right)_{O1'2'3'} = -\nabla' p + \rho \left[\underset{(1)}{\underline{g}} - \underset{(2)}{\frac{d\underline{U}}{dt}} - \underset{(3)}{2\underline{\Omega} \times \underline{u}'} - \underset{(4)}{\frac{d\underline{\Omega}}{dt} \times \underline{x}' - \underline{\Omega} \times (\underline{\Omega} \times \underline{x}')} \right] + \mu \nabla'^2 \underline{u}'$$

(1) (2) (3) (4)

provides the incompressible NS equations in non-inertial reference frame where ' denotes differentiation, velocity, and position in the O' reference frame.

[] terms are body forces due to the motion of O'

pressure & viscous stresses are independent of reference frame

Thermodynamic properties & vectors & vector operators $\nabla, \nabla', \partial_x, \partial'_x$ (relative)

but not their components
For $\underline{U} = \text{constant}$ & $\underline{\Omega} = 0$: [] = \underline{g} i.e. weight in reference frame

for \underline{U} and $\underline{\Omega} = \text{constant}$ at $\underline{u}' = 0$: [] = $\underline{g} - \underline{\dot{U}} - \underline{\dot{\Omega}} \times (\underline{\Omega} \times \underline{x}')$

rigid body translation or rotation

Such that $\rho \nabla'^2 \underline{u}' = 0$ & $\nabla' p = \rho(\underline{g} - \underline{a})$

- ① $\ddot{\underline{r}}$ = acceleration O' relative to O
- ② $-2\Omega \times \underline{u}' = \text{Coriolis acceleration} = f(\underline{u}') \neq f(\underline{x}')$
- ③ $-\dot{\underline{\Omega}} \times \underline{x}' = f(\dot{\underline{\Omega}})$ i.e. rate of change $\underline{\Omega}$
- ④ $-\underline{\Omega} \times (\underline{\Omega} \times \underline{x}') = \text{centrifugal acceleration } f(\underline{\Omega}, \underline{x}')$

① Apparent force pushes person back in seat or tighten grip hand rail when accelerating.
 Aircraft parabolic trajectory weepers inferior when $\dot{v} = g$.

② $f(\underline{u}')$ not \underline{x} . Important navigation (air/sea) ^{artillery} at Λ

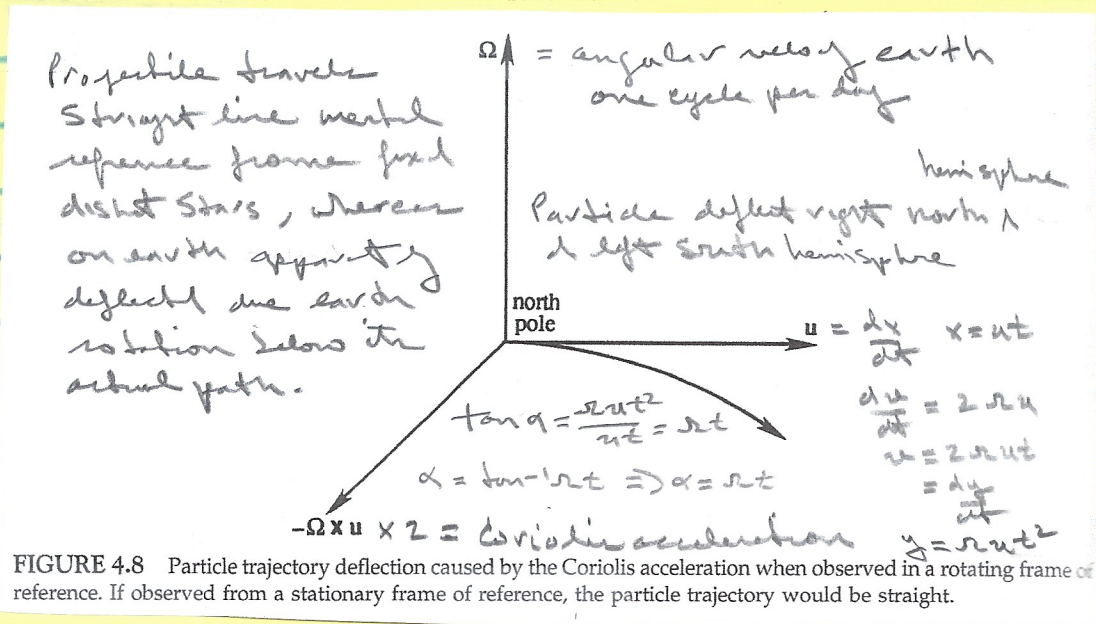


FIGURE 4.8 Particle trajectory deflection caused by the Coriolis acceleration when observed in a rotating frame of reference. If observed from a stationary frame of reference, the particle trajectory would be straight.

Also important geophysical fluid dynamics

③ Important when $\underline{r}(t)$ or direction rotation change with time

④ Centrifugal acceleration = $f(\underline{\Omega}, \underline{x}')$

↑

Consider: $\underline{\Omega} = (0, 0, \Omega)$

↑
define axis rotation

$\underline{x}' = (R, \alpha, z)$

$$-\underline{\Omega} \times (\underline{\Omega} \times \underline{x}') = \Omega^2 R \hat{e}_r$$

= apparent acceleration

$$\underline{g} = \underline{g}_n + \Omega^2 R \hat{e}_r = \text{effective acceleration}$$

↑
Newtonian \underline{g}_n towards earth center

$$\underline{g}_n = -\nabla \Phi$$

$$\underline{g} = g \hat{e}_z$$

Body force potential can be found for new form, but only input of large atmospheric or oceanic scale flows: $-\frac{1}{2} \rho \Omega^2 (x^2 + y^2)$

Earth surface
ellipsoidal with
equatorial Δ
42 km larger
polar diameter

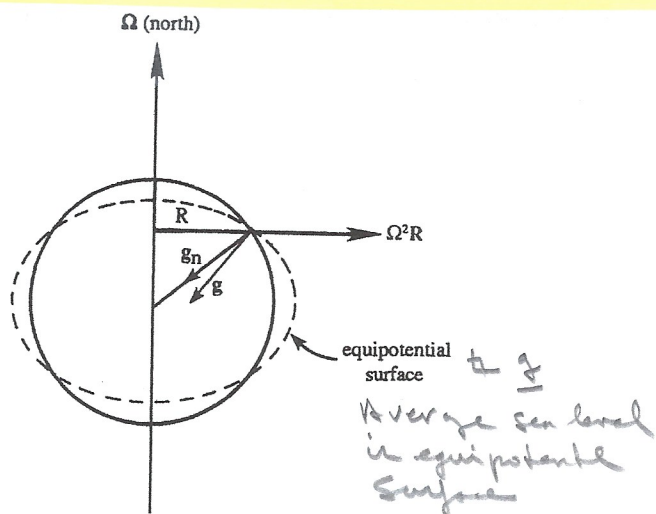


FIGURE 4.9 The earth's rotation causes it to bulge near the equator and this leads to a mild distortion of equipotential surfaces from perfect spherical symmetry. The total gravitational acceleration is a sum of a centrally directed acceleration g_n (the Newtonian gravitation) and a rotational correction $\Omega^2 R$ that points away from the axis of rotation.

$$\underline{u} = \frac{d\underline{x}}{dt} = \frac{d\underline{x}}{dt} + \frac{d\underline{x}'}{dt} = \underline{u} + \frac{d}{dt} (x_1' \underline{e}_1 + x_2' \underline{e}_2 + x_3' \underline{e}_3)$$

$$= \underline{u} + \underline{u}' + \underline{\Omega} \times \underline{x}'$$

$$\underline{u}' = \dot{x}_1' \underline{e}_1 + \dot{x}_2' \underline{e}_2 + \dot{x}_3' \underline{e}_3$$

$$\underline{\Omega} \times \underline{x}' = x_1' \dot{\underline{e}}_1 + x_2' \dot{\underline{e}}_2 + x_3' \dot{\underline{e}}_3$$

$$\frac{d\underline{x}'}{dt} = \underline{u}' + \underline{\Omega} \times \underline{x}'$$

$$\frac{d\underline{u}}{dt} = \frac{d\underline{u}}{dt} + \frac{d\underline{u}'}{dt} + \frac{d}{dt} (\underline{\Omega} \times \underline{x}')$$

$$= \frac{d\underline{u}}{dt} + \frac{d}{dt} (u_1' \underline{e}_1 + u_2' \underline{e}_2 + u_3' \underline{e}_3) + \frac{d\underline{\Omega}}{dt} \times \underline{x}' + \underline{\Omega} \times \frac{d\underline{x}'}{dt}$$

$$= \frac{d\underline{u}}{dt} + \underbrace{\frac{du_1'}{dt} \underline{e}_1 + \frac{du_2'}{dt} \underline{e}_2 + \frac{du_3'}{dt} \underline{e}_3}_{\underline{u}'} + \underbrace{u_1' \dot{\underline{e}}_1 + u_2' \dot{\underline{e}}_2 + u_3' \dot{\underline{e}}_3}_{\underline{\Omega} \times \underline{u}'} + \underline{\Omega} \times (\underline{u}' + \underline{\Omega} \times \underline{x}')$$

$$= \frac{d\underline{u}}{dt} + \underline{u}' + 2 \underline{\Omega} \times \underline{u}' + \frac{d\underline{\Omega}}{dt} \times \underline{x}' + \underline{\Omega} \times (\underline{\Omega} \times \underline{x}')$$

material frame WS: $\rho \frac{D\underline{u}}{Dt} = -\nabla p + \underline{e}_g + \mu \nabla^2 \underline{u}$

$$\stackrel{\circ}{\circ} \quad \frac{D\underline{u}'}{Dt} = \frac{D\underline{u}'}{Dt} + \underline{\dot{\Omega}} : + 2 \underline{\Omega} \times \underline{u}' + \frac{d\underline{\Omega}}{dt} \times \underline{x}' + \underline{\Omega} \times (\underline{\Omega} \times \underline{x}')$$

$$= -\nabla p / \rho + \underline{g} + \nu \nabla^2 \underline{u}'$$

$$\circ \quad \rho \frac{D\underline{u}'}{Dt} = -\nabla p + \rho \left[\underline{g} - \underline{\dot{\Omega}} - 2 \underline{\Omega} \times \underline{u}' + \frac{d\underline{\Omega}}{dt} \times \underline{x}' + \underline{\Omega} \times (\underline{\Omega} \times \underline{x}') \right]$$

$$+ \mu \nabla^2 \underline{u}'$$

Exercise 4.50. Derive (4.43) from (4.42).

Solution 4.50. The entire statement of equation (4.42) is:

$$\begin{aligned}\mathbf{u} &= \frac{d\mathbf{x}}{dt} = \frac{d\mathbf{X}}{dt} + \frac{d\mathbf{x}'}{dt} = \mathbf{U} + \frac{d}{dt}(x'_1\mathbf{e}'_1 + x'_2\mathbf{e}'_2 + x'_3\mathbf{e}'_3) \\ &= \mathbf{U} + \frac{dx'_1}{dt}\mathbf{e}'_1 + \frac{dx'_2}{dt}\mathbf{e}'_2 + \frac{dx'_3}{dt}\mathbf{e}'_3 + x'_1\frac{d\mathbf{e}'_1}{dt} + x'_2\frac{d\mathbf{e}'_2}{dt} + x'_3\frac{d\mathbf{e}'_3}{dt} = \mathbf{U} + \mathbf{u}' + \boldsymbol{\Omega} \times \mathbf{x}' ,\end{aligned}$$

Thus, we see that $d\mathbf{x}'/dt = \mathbf{u}' + \boldsymbol{\Omega} \times \mathbf{x}'$. Now time differentiate \mathbf{u} , to find:

$$\begin{aligned}\frac{d\mathbf{u}}{dt} \equiv \mathbf{a} &= \frac{d\mathbf{U}}{dt} + \frac{d\mathbf{u}'}{dt} + \frac{d}{dt}(\boldsymbol{\Omega} \times \mathbf{x}') \\ &= \frac{d\mathbf{U}}{dt} + \frac{d}{dt}(u'_1\mathbf{e}'_1 + u'_2\mathbf{e}'_2 + u'_3\mathbf{e}'_3) + \frac{d\boldsymbol{\Omega}}{dt} \times \mathbf{x}' + \boldsymbol{\Omega} \times \frac{d\mathbf{x}'}{dt} \\ &= \frac{d\mathbf{U}}{dt} + \frac{du'_1}{dt}\mathbf{e}'_1 + \frac{du'_2}{dt}\mathbf{e}'_2 + \frac{du'_3}{dt}\mathbf{e}'_3 + u'_1\frac{d\mathbf{e}'_1}{dt} + u'_2\frac{d\mathbf{e}'_2}{dt} + u'_3\frac{d\mathbf{e}'_3}{dt} + \frac{d\boldsymbol{\Omega}}{dt} \times \mathbf{x}' + \boldsymbol{\Omega} \times (\mathbf{u}' + \boldsymbol{\Omega} \times \mathbf{x}')\end{aligned}$$

Here the various terms written in component form may be identified. The second through fourth terms are the fluid particle acceleration, \mathbf{a}' , observed in the non-inertial frame of reference. The fifth through seventh terms, which involve the time derivatives of the unit vectors, can be written in terms of a cross product:

$$u'_1\frac{d\mathbf{e}'_1}{dt} + u'_2\frac{d\mathbf{e}'_2}{dt} + u'_3\frac{d\mathbf{e}'_3}{dt} = \boldsymbol{\Omega} \times \mathbf{u}' ,$$

as depicted in Figure 4.7 and described in paragraph below (4.42). With these replacements, the last equality for the fluid particle acceleration becomes:

$$\begin{aligned}\mathbf{a} &= \frac{d\mathbf{U}}{dt} + \mathbf{a}' + \boldsymbol{\Omega} \times \mathbf{u}' + \frac{d\boldsymbol{\Omega}}{dt} \times \mathbf{x}' + \boldsymbol{\Omega} \times \mathbf{u}' + \boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{x}') \\ &= \frac{d\mathbf{U}}{dt} + \mathbf{a}' + 2\boldsymbol{\Omega} \times \mathbf{u}' + \frac{d\boldsymbol{\Omega}}{dt} \times \mathbf{x}' + \boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{x}') ,\end{aligned}$$

which matches (4.43).

Right Circular Cone

Let R = radius of base, s = slant height.

$$s = \sqrt{R^2 + h^2}$$

$$S = \pi R s = \pi R \sqrt{R^2 + h^2}$$

$$T = \pi R(R + s) = \pi R(R + \sqrt{R^2 + h^2})$$

$$V = \frac{1}{3}\pi R^2 h$$

