

Axially Symmetric Wake

$$u_1(x, r) = U_\infty - u(x, r) \ll U_\infty \quad \text{far downstream}$$

$$u = U - u_1 \quad v = w \ll U \quad \text{so neglect higher order terms}$$

$$u_x = -u_{1x}$$

$$u_v = -u_{1v}$$

$$u_{vv} = -u_{1vv}$$

$$u u_x + v u_v = \frac{\nu}{r} (r u_v)_r = \nu (u_{1v}/r + u_{1vv})$$

$$(U - u_1)(-u_{1x}) + u_1(-u_{1v}) = \nu (-u_{1v}/r - u_{1vv})$$

$$U u_{1x} = \frac{\nu}{r} \frac{\partial}{\partial r} (r u_{1v}) = \nu (u_{1v}/r + u_{1vv})$$

$$D = \int_0^\infty \int_0^\infty u(U-u) r dr d\theta = 2\pi \int_0^\infty u_1(U-u_1) r dr$$

$$= 2\pi U \int_0^\infty u_1 r dr = \text{constant}$$

$$\text{so } u_1 = c U f(\eta) / x = a(x) f$$

analogous 2D

$$\eta = \frac{1}{2} r \sqrt{\frac{U}{\nu x}} = r b(x) \quad r = \eta / b(x)$$

$$r = \eta / b(x)$$

$$dr = d\eta / b \quad r dr = \frac{\eta d\eta}{b^2}$$

$$D = 2\pi U \int_0^\infty a(x) f(\eta) r dr$$

$$= 2\pi U \frac{a(x)}{b(x)^2} \int_0^\infty f(\eta) \eta d\eta \quad a(x) = \frac{cU}{x}$$

$$= 2\pi U \frac{cU}{x} \frac{1}{b} \int_0^\infty f(\eta) \eta d\eta \quad b = \left[\frac{\nu}{4\nu x} \right]^{1/2} \quad b^2 = \frac{\nu}{4\nu x}$$

$$= 8\pi U c U \int_0^\infty f(\eta) \eta d\eta = f(x)$$

$$u_{1x} = a_x f + a f' \gamma_x$$

$$u_{1v} = a f' \gamma_v$$

$$u_{1vv} = a f'' \gamma_v^2$$

$$\gamma_x = v \delta_x = \gamma b_x / b$$

$$\gamma_v = b = \frac{1}{2} (\sigma/v)^{1/2} x^{-1/2}$$

$$\delta_x = -\frac{1}{4} (\sigma/v)^{1/2} x^{-3/2}$$

$$a = c v x^{-1}$$

$$a_x = -c v x^{-2}$$

$$\gamma_{v/v} = b^2 / \gamma$$

$$U(a_x f + a f' \gamma_x)$$

$$= v (a f' \gamma_{v/v} + a f'' \gamma_v^2)$$

$$v a f'' \gamma_v^2 + v a f' \gamma_{v/v} - U a_x f - U a f' \gamma_x = 0$$

$$f'' (v a \gamma_v^2) + f' (v a \gamma_{v/v} - U a \gamma_x) - U a_x f = 0$$

$$f'' (v a b^2) + f' (v a b^2 / \gamma - U a \gamma \delta_x / b) - U a_x f = 0$$

$$v a b^2 = v c v x^{-1} \frac{\sigma}{4 v x} = c v^2 / 4 x^2$$

$$b_x / b = -\frac{1}{4} (\frac{\sigma}{v})^{1/2} x^{-3/2}$$

$$v a b^2 / \gamma = c v^2 / 4 x^2 \gamma$$

$$\times 2 (\frac{v}{\sigma})^{1/2} x^{1/2}$$

$$-U a \gamma \delta_x / b = -U \gamma (c v x^{-1}) (-\frac{1}{2} x^{-1})$$

$$= -\frac{1}{2} x^{-1}$$

$$= U^2 c \gamma / 2 x^2$$

$$-U a_x = -U (-c v x^{-2}) = U^2 c / x^2$$

$$f'' (c v^2 / 4 x^2) + f' (c v^2 / 4 x^2 \gamma + U^2 c \gamma / 2 x^2) + U^2 c / x^2 f$$

$$f'' + f' (\gamma^{-1} + 2\gamma) + f = 0$$

$$\gamma f'' + f' + 2\gamma^2 f' + \gamma f = 0$$

$$\underbrace{(\gamma f')}'$$

$$(\gamma f')' + 2\gamma^2 f' + 4\gamma f = 0$$

$$u_{1y}(0) = 0 \Rightarrow f'(0) = 0$$

$$u_{1y}(\infty) = 0 \Rightarrow f(\infty) = 0$$

$$f(\eta) = \exp(-\eta^2)$$

$$f' = -2\eta \exp(-\eta^2)$$

$$f'' = 4\eta^2 \exp(-\eta^2) - 2e^{-\eta^2}$$

OK

$$4\eta^3 e^{-\eta^2} - 2\eta e^{-\eta^2} - 2\eta e^{-\eta^2} - 4\eta^3 e^{-\eta^2} + 4\eta e^{-\eta^2} = 0$$

$$u_1 = \frac{C_0 \sigma}{x} e^{-\eta^2} = \eta^2 = \frac{\sigma}{4\nu x} r^2 \quad \eta = \left[\frac{\sigma}{4\nu x} \right]^{1/2} r$$

$$v = \left[\frac{4\nu x}{\sigma} \right]^{1/2} \eta \propto x^{1/2}$$

$$D = 8\pi \rho \sigma \nu \int_0^{\infty} e^{-\eta^2} \eta d\eta = 8\pi \rho \sigma \nu \left[\frac{1}{2} \right] = 4\pi \rho \sigma \nu$$

$$\int_0^{\infty} x^n e^{-ax^2} dx = \frac{\Gamma\left(\frac{n+1}{2}\right)}{2(a^{n+1/2})} \quad n > -1$$

$$a=1$$

$$n=1$$

$$= \frac{\Gamma(1)}{2} = \frac{1}{2}$$

$$C_0 = 4\pi \rho \sigma \nu / \frac{1}{2} \rho \sigma^2 A_p$$

$$= 8\pi \rho \nu / \sigma A_p$$

$$C = C_0 \sigma A_p / 8\pi \nu$$

$A_p =$
projected
area

$$\Gamma(n) = (n-1)! \quad (0)! = 1$$

$$(1)! = 1$$

$$u_1 = \frac{C_0 \sigma L^2}{8\pi \nu} \frac{\sigma}{x} e^{-\eta^2}$$

$$A_p = \frac{\pi D^2}{4}$$

$$C = \frac{C_0 \sigma \pi D^2 / 4}{8\pi \nu} = \frac{C_0 \rho \sigma D}{32}$$

$$Re = \sigma D / \nu$$

$$u_1 / \sigma = \frac{C_0}{8\pi} (Re) \frac{L}{x} \exp\left(-\frac{\sigma r^2}{4x}\right) \quad u_{max} \propto x^{-1}$$

$$u_1 / \sigma = \frac{C_0}{32} (Re) \frac{D}{x} e^{-\eta^2}$$

same shape as wake with $y=r$

$$Re = \sigma D / \nu$$

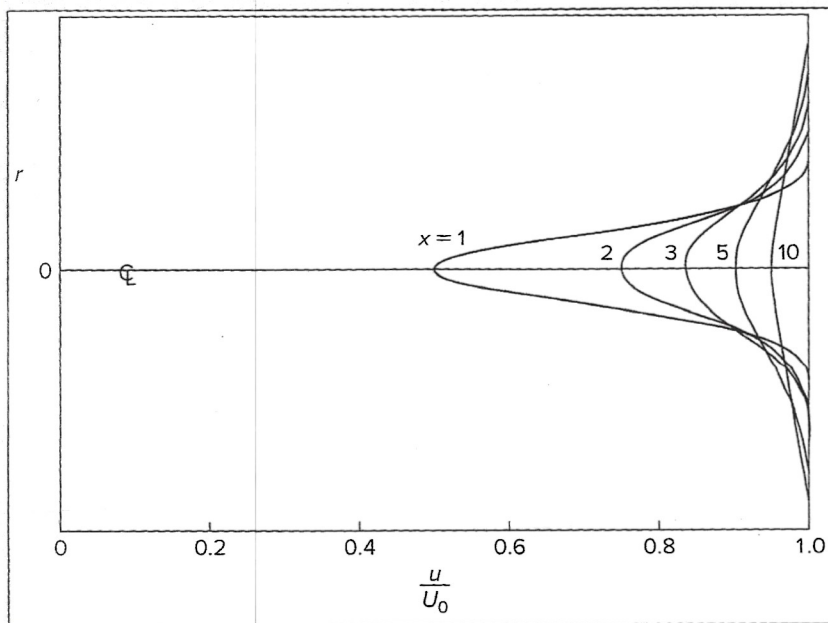


FIGURE 4-43

Velocity profiles illustrating the decay of a round wake from Eq. (4-232). Units on x and r are arbitrary.