

2D Far Wake of Nonlifting Bodies

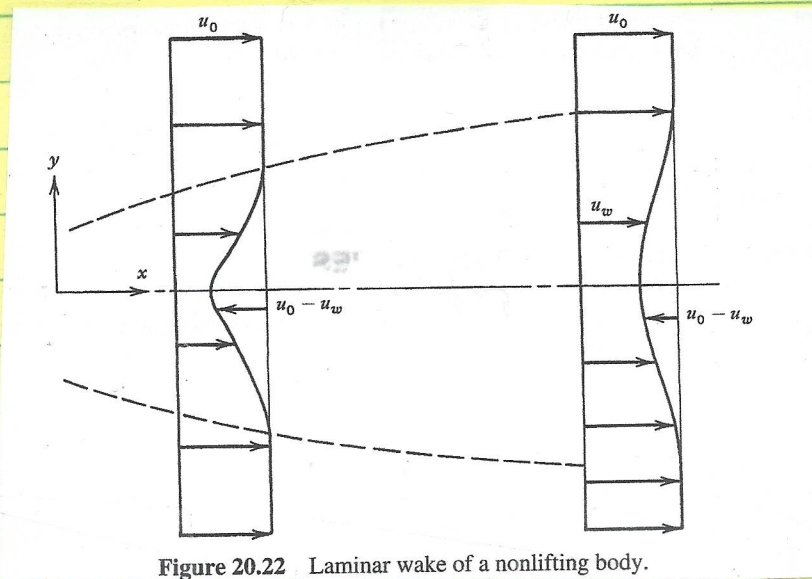


Figure 20.22 Laminar wake of a nonlifting body.

Assume: $Re_{Body} = \frac{u_0 b}{\nu}$ large, similarity for the velocity defect $u = u_0 - u_w$, & BL equations with $p_x = 0$.

$$u_x + v_y = 0$$

$$u u_x + v u_y = \nu u_{yy}$$

Subscript w not shown

$$u_w = u_0 - u \quad v_w = v$$

$$u = u_0 - u_w$$

$$u_{wx} = -u_x$$

= velocity defect

$$u_{wy} = -u_y$$

$$u_{wyy} = -u_{yy}$$

$$(u_0 - u)(-u_x) + v(-u_y) = -\nu(-u_{yy})$$

$$(u_0 - u)u_x + v u_y = \nu u_{yy}$$

Assume $u = \gamma y$ $v = -x x$

$$x = A x^p f(\gamma) \quad \gamma = y / B x^q \quad \gamma y = y^{-1}$$

$$= a(x) f(\gamma) \quad = y / b(x) \quad \gamma x = -\gamma y \frac{x}{y}$$

A, B, p, q chosen as part of solution.

$$u = \frac{a}{b} f' \quad v = -axf + f' (\gamma a b x / b)$$

$$u_x = (a b^{-1})_x f' + (a b^{-1}) f'' (-\gamma b x / b)$$

$$u_y = (a b^{-2}) f''$$

$$u_{yy} = (a b^{-3}) f'''$$

$$a_x = A p x^{p-1}$$

$$b_x = B q x^{q-1}$$

$$a_x b = A B p x^{p+q-1}$$

$$\gamma_x = A B q x^{p+q-1}$$

$$u_0 u_x = u_0 [(a_x b^{-1} - a b^{-2} b_x) f' - (\gamma a b x / b^2) f'']$$

$$= u_0 [(A p x^{p-1} / B x^q - A B q x^{p+q-1} / B^2 x^{2q}) f' - (\gamma A B q x^{p+q-1} / B^2 x^{2q}) f'']$$

$$= u_0 [(\frac{A p}{B} x^{p-q-1} - \frac{A q}{B} x^{p-q-1}) f' - \gamma \frac{A q}{B} x^{p-q-1} f'']$$

$$u_0 u_x = u_0 \frac{A}{B} x^{p-q-1} [(p-q) f' - \gamma q f'']$$

$$-u u_x + v u_y = - [(A x^p / B x^q) f'] [\frac{A}{B} x^{p-q-1} ((p-q) f' - \gamma q f'')]$$

$$(-A p x^{p-1} f + f' (\gamma A B q x^{p+q-1} / B x^q)) (\frac{A x^p}{B^2 x^{2q}} f'')$$

$$- \frac{A^2}{B^2} x^{2p-2q-1} [(p-q) f'^2 + \gamma q f' f''] \quad \frac{A}{B^2} x^{p-2q}$$

$$- \frac{A^2 p}{B^2} x^{2p-2q-1} f f'' + \gamma \frac{A^2 q}{B^2} x^{2p-2q-1} f' f''$$

$$- \frac{A^2}{B^2} x^{2p-2q-1} [(p-q) f'^2 - 2\gamma q f' f'' + p f f'']$$

$$u_{yy} = (A x^p / B^3 x^{3q}) f''' = \frac{A}{B^3} x^{p-3q} f'''$$

$$u_0 u_x - u u_x + u_0 u_y = \nu u_{yy}$$

$$u_0 B^2 x^{2\beta-1} [\textcircled{1}] + \frac{B}{A} x^{p+\beta-1} [-\textcircled{2}] = \nu f''$$

$$-\textcircled{2} = (\beta-p) f'^2 + 2\gamma f f'' - p f f''$$

$$F_D = \rho u_0^2 \Theta = \rho u_0^2 \int_{-\infty}^{\infty} \left[\frac{u_0}{u_0} - \left(\frac{u_0}{u_0} \right)^2 \right] dy$$

$$u = (a\beta^{-1}) f' \quad = \rho u_0^2 \int_{-\infty}^{\infty} u_0^{-2} [u_0 u_0 - u_0^2] dy$$

$$(u_0 - (a\beta^{-1}) f') (a\beta^{-1}) f'$$

$$u_0 (a\beta^{-1}) f' - (a\beta^{-1})^2 f'^2 = \rho \int_{-\infty}^{\infty} u_0 (u_0 - u) dy$$

$$\eta = y \beta^{-1}$$

$$\frac{u_0 A}{B} x^{p-\beta} f' - \frac{A^2}{B^2} x^{2p-2\beta} f'^2$$

$$= \rho \int_{-\infty}^{\infty} (u_0 - u)(u) dy$$

$$d\eta = dy \beta^{-1}$$

$$B \times \beta dy = dy$$

$$\times B x^\beta = u_0 A x^p f' - \frac{A^2}{B} x^{2p-\beta} f'^2$$

$p=0 \quad \beta = \frac{1}{2}$ underest
 $x \sim x \rightarrow \infty$

$$F_D = \rho A x^p \int_{-\infty}^{\infty} \left[u_0 - \frac{A}{B} x^{p-\beta} f' \right] f' dy$$

$$\frac{u_0 B^2}{2} [-f' - \gamma f''] + \frac{B}{A} x^{-1/2} [-\textcircled{2}] = \nu f''' \quad \text{Puls highlight}$$

= nonlinear

Terms drop out
seen approximation

$\lim_{x \rightarrow \infty}$

$$f''' + \frac{u_0 B^2}{2} (f' + \gamma f'') = 0$$

$$F_D = \rho A u_0 \int_{-\infty}^{\infty} f' dy$$

$$a(x) = A x^p = A$$

$$b(x) = B x^{1/2}$$

$$d\eta = dy / B x^{1/2}$$

$$u = \frac{a}{b} f'$$

$$f' = \frac{b u}{a} = \frac{B x^{1/2}}{A} u$$

$$Q = \int u \cdot n \cdot dA$$

$$= \int u dy$$

per unit span

$$= \rho A u_0 \int_{-\infty}^{\infty} \frac{B x^{1/2}}{A} u \frac{dy}{B x^{1/2}}$$

$$= \rho u_0 \int_{-\infty}^{\infty} u dy$$

$$= \rho u_0 Q$$

$$B = \left(\frac{\nu}{u_0}\right)^{1/2} \quad B^2 = \frac{\nu}{u_0} \times \frac{u_0}{2\nu} = 2 \Rightarrow f''' + 2(f' + \eta f'') = 0$$

$$f' = e^{-\eta^2} \quad f'' = -2\eta e^{-\eta^2} \quad \frac{d}{dx} e^u = e^u \frac{du}{dx}$$

$$f''' = -2e^{-\eta^2} + 4\eta^2 e^{-\eta^2}$$

$$2(f' + \eta f'') = 2(e^{-\eta^2} - 2\eta^2 e^{-\eta^2}) \quad \text{OK}$$

$$F_0 = \rho A u_0 \int_{-\infty}^{\infty} e^{-\eta^2} d\eta = \rho A u_0 \pi^{1/2} \Rightarrow A = \frac{F_0}{\rho u_0 \sqrt{\pi}}$$

$$\eta = y / \left(\frac{\nu}{u_0}\right)^{1/2} x^{1/2} = y \sqrt{\frac{u_0}{4\nu x}}$$

$$\chi = \frac{F_0}{\rho u_0 \pi^{1/2}} \underbrace{\int e^{-\eta^2} d\eta}_{\frac{\sqrt{\pi}}{2} \operatorname{erf}(\eta)}$$

$$= \frac{2F_0}{\rho u_0} \operatorname{erf}(\eta)$$

$$\frac{d\chi}{d\eta} = e^{-\eta^2}$$

$$\chi = \int_0^{\eta} e^{-\eta^2} d\eta$$

$$\operatorname{erf}(\eta) = \frac{2}{\sqrt{\pi}} \int_0^{\eta} e^{-\eta^2} d\eta$$

$$u = u_0 - uw = \frac{A}{B} e^{-\eta^2} = \frac{F_0}{\rho u_0 \sqrt{\pi}} \frac{e^{-\eta^2}}{\left(\frac{\nu x}{u_0}\right)^{1/2}}$$

$$u = \left[\frac{F_0^2}{\rho^2 u_0^2 \pi} \times \frac{u_0}{\nu x} \right]^{1/2} e^{-\eta^2}$$

$$u = \left[\frac{F_0^2}{4\pi \rho^2 u_0 \nu x} \right]^{1/2} e^{-\eta^2}$$

velocity defect
decays $x^{-1/2}$

$$y = B x^{1/2} \eta = \left(\frac{\nu}{u_0}\right)^{1/2} x^{1/2} \eta \quad \text{grows } x^{1/2}$$

Momentum thickness wake = constant = $\theta = F_0 / \rho u_0^2$

whereas in the far wake $\theta = F_0 / \rho u_0 = u_0 \theta$

↑ velocity deft u/s

It can be shown that $\theta = u_0 \delta^*$ thus $\delta^* = \theta$
in the far wake

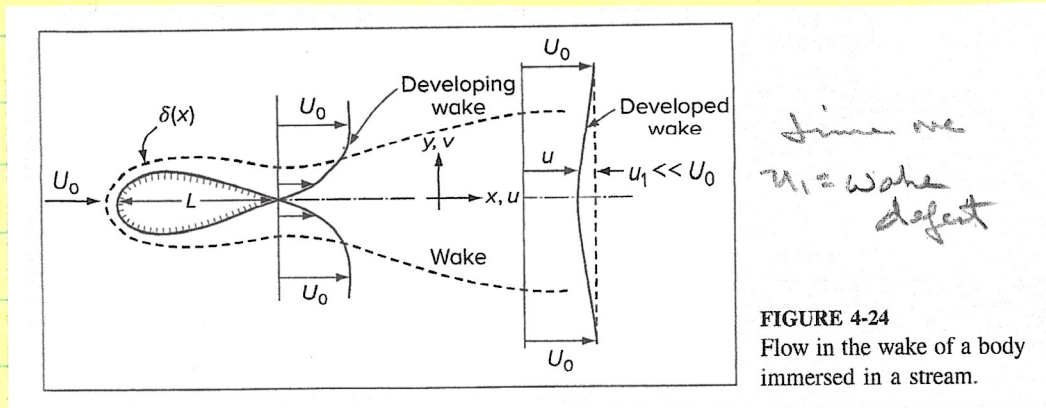
Note we assumed large Re but additionally require $x \rightarrow \infty$ i.e. far wake. It is interesting to note that the same result is obtained using the Oseen equation, which are derived for low Re i.e. laminar convection.

$$\int_{\delta^*}^{\infty} u_0 dy = \int_0^{\infty} u_0 dy = \int_0^{\infty} u_0 dy - \int_0^{\delta^*} u_0 dy$$

$$\text{i.e. } u_0 \delta^* = \int_0^{\infty} (u_0 - u) dy = \int_0^{\infty} u dy = \theta \quad \begin{array}{l} \text{velocity} \\ \text{deficit} \end{array}$$

i.e. in the far wake the velocity deficit momentum & displacement thickness are equal

Plane laminar wake: far-field approximation



$x > 3L$

wake attains similarity

$u_1(x, y) = U_0 - u(x, y) \ll U_0$

BL assumptions: $u u_x + v u_y = \nu u_{yy}$ with $p_x = 0$

Also Oseen approximation, i.e., linearized

convection: $U_0 u_{1x} = \nu u_{1yy}$ $u_1(x, \pm\infty) = 0$
 $u_{1y}(x, 0) = 0$

Use the jet plate far wake solution

$u_1 = c U_0 \left(\frac{L}{x}\right)^{1/2} e^{-\gamma^2/4}$ $\gamma = y \left(\frac{U_0}{\nu x}\right)^{1/2}$

per unit depth

$F = \int_{-\infty}^{\infty} c U_0 u_1 dy = c U_0 c U_0 \left(\frac{L}{x}\right)^{1/2} \int_{-\infty}^{\infty} e^{-\gamma^2/4} d\gamma \left(\frac{\nu x}{U_0}\right)^{1/2}$
 $= c U_0^2 c \left(\frac{L \nu x}{x U_0}\right)^{1/2} \int_{-\infty}^{\infty} e^{-\gamma^2/4} d\gamma$

$\frac{1}{2} c_0 \frac{U_0^2}{U_0} L = 2 c U_0^2 c \left(\frac{L \nu \pi}{U_0}\right)^{1/2} 2\pi^{1/2}$

$\frac{1}{4} \left(\frac{U_0}{L \nu \pi}\right)^{1/2} c_0 L = c = \left(\frac{U_0 L}{16 \nu \pi}\right)^{1/2} c_0 = c_0 \left(\frac{Re_L}{16 \pi}\right)$

$u_1/U_0 = c_0 \left(\frac{Re_L}{16 \pi}\right)^{1/2} \left(\frac{L}{x}\right)^{1/2} e^{-\gamma^2/4}$ $\propto c_0$

valid $x > 3L$