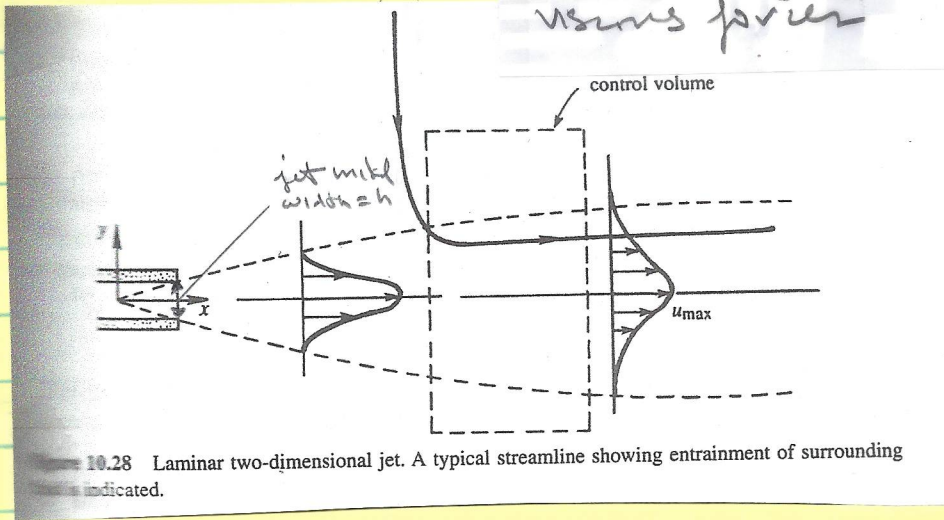


# Plane Laminar Jet

( $v \neq 0$  due to entrainment)  
 Entrainment = jet entrains surrounding fluid due to viscous forces



10.28 Laminar two-dimensional jet. A typical streamline showing entrainment of surrounding fluid is indicated.

near field depends  $Re$ , whereas far field achieves similarity due to absence of externally imposed length scale in downstream direction

BL assumptions:  $Re = \frac{u_0 x}{\nu} \gg 1$   $u_0 = \text{uniform inlet velocity}$

$\frac{\partial}{\partial y} \gg \frac{\partial}{\partial x}$   $\nu \ll x$   $\frac{\partial p}{\partial x} = \frac{\partial p}{\partial y} = 0$  Note:  $Re_{crit} = 4$

$\frac{d}{dy}(uv) = \nu u_{yy} + \nu v_{yy}$

$u_x + v_y = 0$   
 $\nu u_x + \nu v_y = \nu u_{yy}$

(1)  $u(\pm\infty) = 0$   
 (2)  $v(\pm\infty) = 0$   
 $u(x=0) = u_0$  for  $-\frac{h}{2} \leq y \leq \frac{h}{2}$   
 $v_y(\pm\infty) = 0$   
 Symmetry Condition

Similar derivation momentum integral equation #3 except  $\int_{-\infty}^{\infty}$

$\int_{-\infty}^{\infty} [u(1) + (2)] dy =$   
 $\int_{-\infty}^{\infty} 2u u_x dy + \int_{-\infty}^{\infty} (\nu u_y + \nu v_y) dy = \nu u_y \Big|_{-\infty}^{\infty} = \frac{\nu}{2} \Big|_{-\infty}^{\infty} 2u_{yy}$   
 $\frac{d}{dx} \int_{-\infty}^{\infty} u^2 dy + \nu \Big|_{-\infty}^{\infty} = \nu u_y \Big|_{-\infty}^{\infty}$   
 $u(\pm\infty) = 0$   
 $u_y(\pm\infty) = 0$

Streamwise momentum flux preserved i.e. zero drag (zero drag & free stream velocity version BL equation) (CV plate plate)

$$\int_{-\infty}^{\infty} u^2 dy = \text{Constant} = \int_{-\infty}^{\infty} \tilde{u}^2(y) dy = \mathcal{J}/\rho$$

$\mathcal{J}$  = momentum flux per unit span = constant

where  $u = \tilde{u}(y)$  at  $x = x_0$   $\tilde{u}(y)$  = known inlet velocity profile

Similarity solution far enough downstream  $x_0$  where BL equations valid at  $\tilde{u}(y)$  w/o influence. Since BL equation same Blasius but expect BL seek solution of same form:

$$\chi = u_0(x) \delta(x) f(\eta) \quad \eta = y/\delta(x) \quad \delta(x) = [\nu x / u_0]^{1/2}$$

$$u_0(x) = u(x, 0) = u_{max}$$

$$u = \chi_y = u_0 \delta f' \eta_y = u_0 f' \quad u(0) = u_0 \Rightarrow f'(0) = 1$$

$$\mathcal{J}/\rho = \int_{-\infty}^{\infty} u^2 dy = u_0^2(x) \int_{-\infty}^{\infty} f'^2(\eta) dy \quad dy = d\eta/\delta$$

$$= u_0^2 \delta \int_{-\infty}^{\infty} f'^2(\eta) d\eta = u_0^2 \delta C \quad C = \text{non dimensional constant}$$

$$= u_0^2 [\nu x / u_0]^{1/2} C$$

$$\mathcal{J}/\rho = C u_0^{3/2} (\nu x)^{1/2} \quad \frac{\mathcal{J}^2}{\rho^2} = C^2 u_0^3 \nu x$$

$$u_0^3 = \mathcal{J}^2 / \rho^2 C^2 \nu x$$

$$u_0(x) = [\mathcal{J}^2 / \rho^2 C^2 \nu x]^{1/3} \quad \delta(x) = [C \rho \nu^2 x^2 / \mathcal{J}]^{1/3}$$

$$(u_0 \delta)^3 = \frac{\mathcal{J}^2}{\rho^2 \nu x} \times \frac{C \rho \nu^2 x^2}{\mathcal{J}} = \frac{\mathcal{J} \nu x}{\rho C}$$

$$\frac{\nu x (C^2 \nu x)^{1/3}}{\mathcal{J}^{2/3}}$$

$$\frac{\nu^{4/3} C^{2/3} x^{4/3}}{\mathcal{J}^{2/3}}$$

$$\chi = \underbrace{\left[ \frac{\mathcal{J} \nu x}{\rho C} \right]^{1/3}}_{a(x)} f(\eta) \quad \eta = y / \underbrace{\left[ \frac{C \rho \nu^2 x^2}{\mathcal{J}} \right]^{1/3}}_{\delta(x)} \quad \mathcal{J}^{2/3}$$

$$\gamma_y \gamma_{yx} - \gamma_x \gamma_{yy} = v \gamma_{yyy}$$

$$\begin{aligned} \gamma &= a(x) f(y) & \gamma &= y/b(x) & \gamma_y &= b^{-1} & \gamma_{yy} &= 0 \\ \alpha &= \gamma_y = a f' b^{-1} & &= y b^{-1} & \gamma_x &= -y b^{-2} b_x & & \\ &= \frac{a}{b} f' & & & &= -y b_x / b & & \end{aligned}$$

$$\alpha_x = (a b^{-1})_x f' + (a b^{-1}) f'' (-y b_x / b)$$

$$\alpha_y = (a b^{-1}) f'' b^{-1} = (a b^{-2}) f''$$

$$\begin{aligned} \alpha_{yy} &= (a b^{-2}) f''' & \alpha &= -\gamma_x = -[a_x f + a f' (-y b_x / b)] \\ & & &= -a_x f + f' (y a b_x / b) \end{aligned}$$

$$\begin{aligned} &(a b^{-1}) f' [(a b^{-1})_x f' + (a b^{-1}) f'' (-y b_x / b)] \\ + &[-a_x f + f' (y a b_x / b)] (a b^{-2}) f'' = v (a b^{-3}) f''' \end{aligned}$$

$$\begin{aligned} &(a b^{-1}) (a b^{-1})_x f'^2 - (a b^{-1})^2 (y b_x / b) f' f'' \\ - &a_x (a b^{-2}) f f'' + (a b^{-2}) (y a b_x / b) f' f'' = v (a b^{-3}) f''' \end{aligned}$$

$$(a b^{-1}) (a b^{-1})_x f'^2 - a_x (a b^{-2}) f f'' - v (a b^{-3}) f''' = 0$$

$$\frac{a}{b} \left( \frac{a_x}{b} - \frac{a}{b^2} b_x \right) f'^2 - \frac{a_x a}{b^2} f f'' - v \frac{a}{b^3} f''' = 0$$

$$b \left( \frac{a_x}{b} - \frac{a}{b^2} b_x \right) f'^2 - a_x a f f'' - v f''' = 0$$

$$(a_x b - a b_x) f'^2 - a_x a f f'' - v f''' = 0$$

$$a = \left[ \frac{J v x}{e^2} \right]^{1/3} = \left( \frac{J v}{e^2} \right)^{1/3} x^{1/3} \quad a_x = \left( \frac{J v}{e^2} \right)^{1/3} \frac{x^{-2/3}}{3}$$

$$b = \left[ \frac{c e v^2 x^2}{J} \right]^{1/3} = \left[ \frac{c e v^2}{J} \right]^{1/3} x^{2/3} \quad b_x = \left[ \frac{c e v^2}{J} \right]^{1/3} \frac{2}{3} x^{-1/3}$$

$$a_x b = \left( \frac{J v}{e^2} \right)^{1/3} \frac{x^{-2/3}}{3} \left[ \frac{c e v^2}{J} \right]^{1/3} x^{2/3}$$

$$= \left( \frac{J v}{e^2} \frac{c e v^2}{J} \right)^{1/3} \frac{1}{3} = \frac{v}{3}$$

$$a > x = \left(\frac{5V}{cL}\right)^{1/3} x^{1/3} \left(\frac{cL^2}{5}\right)^{1/3} \frac{2}{3} x^{-1/3}$$

$$= \left(\frac{5V}{cL} \frac{cL^2}{5}\right)^{1/3} \frac{2}{3} = \frac{2V}{3}$$

$$\left(\frac{V}{3} - \frac{2V}{3}\right) f'^2 - \frac{V}{3} - \sqrt{f} f''' = 0$$

$$3 \frac{d}{d\eta} (f'') + \frac{d}{d\eta} (ff) = 0$$

$$\text{ie } 3f''' + f''f + f'^2 = 0 \quad (1)$$

$$\text{BC } f' = 0 \quad \eta = \pm\infty \quad f' = 1 \quad \eta = 0 \quad f = 0 \quad \eta = 0$$

$$u(x, \pm\infty) = 0$$

$$u(x, 0) = u_0(x)$$

$$u(x, 0) = 0$$

$$\text{Integrating (1): } 3f'' + f'f = c_1 \quad (2) \quad \text{at } \eta = \pm\infty \quad f' = 0 \quad \text{so } f'' = 0$$

$$\text{so } c_1 = 0$$

$$\text{Integrating (2): } 3f' + f^2/2 = c_2 \quad (3)$$

$$3 \frac{d}{d\eta} (f') + \frac{d}{d\eta} (f^2/2) = 0$$

$$\text{at } \eta = 0 \quad f' = 1 \quad f = 0 \Rightarrow c_2 = 3$$

$$\frac{df}{d\eta} + f^2/6 = 1$$

$$df = (1 - f^2/6) d\eta$$

$$3 \frac{df}{d\eta} + f^2/2 = 3 \Rightarrow \int \frac{df}{1 - f^2/6} = \int d\eta$$

$$f = \sqrt{6} \tanh \beta$$

$$df = \sqrt{6} \operatorname{sech}^2 \beta d\beta$$

$$\int \frac{\sqrt{6} \operatorname{sech}^2 \beta d\beta}{1 - \tanh^2 \beta} = \int d\eta$$

$$\beta = \frac{\eta}{\sqrt{6}} + c_3$$

$$\beta = \operatorname{tanh}^{-1} f/\sqrt{6}$$

$$f = \sqrt{6} \tanh \left( \frac{\eta}{\sqrt{6}} + c_3 \right)$$

$$f(0) = 0 \Rightarrow c_3 = 0$$

$$f' = \operatorname{sech}^2 \left( \frac{\eta}{\sqrt{6}} \right)$$

$$u(x, y) = u_0(x) f'(\eta) = \left( \frac{3^2}{c^2 \rho^2 \nu x} \right)^{1/3} \operatorname{sech}^2 \left( \frac{\eta}{\sqrt{6}} \left[ \frac{3}{c \rho \nu x} \right]^{1/3} \right)$$

$$C = \int_{-\infty}^{\infty} f'^2(\eta) d\eta = \int_{-\infty}^{\infty} \operatorname{sech}^4 \left( \frac{\eta}{\sqrt{6}} \right) d\eta = \frac{4\sqrt{6}}{3}$$

$$\dot{m} = \int_{-\infty}^{\infty} \rho u_0(x) f'(\eta) d\eta = \rho u_0(x) \delta(x) \int_{-\infty}^{\infty} f'(\eta) d\eta$$

$$u_0(x) = \left[ \frac{3^2}{c^2 \rho^2 \nu x} \right]^{1/3} \propto x^{-1/3}$$

$$= \rho u_0(x) \delta(x) f \Big|_{-\infty}^{\infty} \quad f = \sqrt{6} \tanh \left( \frac{\eta}{\sqrt{6}} \right)$$

$$= \rho u_0(x) \delta(x) 2\sqrt{6} \quad 1 - (-1) = 2$$

$$\delta(x) = \left[ \frac{c \rho \nu x^2}{3} \right]^{1/3} \propto x^{2/3} \quad (2\sqrt{6})^3 = 4 \times 6 (2\sqrt{6}) = 4 \times 12 \times \sqrt{6}$$

$$u_0 \delta = \left[ \frac{3^2}{c^2 \rho^2 \nu x} \frac{c \rho \nu x^2}{3} \right]^{1/3} = \left[ \frac{3 \nu x}{c \rho} \right]^{1/3} = \left[ \frac{3 \nu x}{4 \rho c} \right]^{1/3}$$

$$\dot{m} = (36 \nu^2 \rho^2 \nu x)^{1/3} \propto x^{1/3}$$

increases downstream due to entrainment

$$v = -\tau_x = -\alpha_x f + f' \left( \eta \frac{3 \nu x}{c \rho} \right)$$

$$= -\frac{1}{3} \left( \frac{3 \nu}{c \rho x^2} \right)^{1/3} f + f' \left( \eta \frac{2 \nu}{3} \left[ \frac{3}{c \rho \nu x} \right]^{1/3} \right)$$

$$= -\frac{1}{3} \left( \frac{3 \nu}{c \rho x^2} \right)^{1/3} [f - 2 \eta f']$$

$$\text{or } v/u_0(x) = -\frac{[f - 2 \eta f']}{3 \sqrt{Re_x}} \quad Re_x = \frac{x u_0(x)}{\nu}$$

$$f(\eta) = \pm \sqrt{6} \quad 2 \eta f'(\eta) = 2 \eta \operatorname{sech}^2(\eta/\sqrt{6}) = 0 \quad \eta = \pm \infty$$

$$\text{or } v/u_0(x) = \pm \frac{\sqrt{6}}{3 \sqrt{Re_x}} \quad \eta = \pm \infty$$

entrainment due to flow from above/below

The jet spreads as  $x \uparrow$ . Using similar  
 when  $Re \downarrow$  at  $x=0$  define half width  
 jet  $h_{99}$  as  $y$  location where  $\eta$  falls  
 to 1%  $y=0$  ( $12.0120(x)$ )

$$\operatorname{sech}^2\left(\frac{h_{99}}{\sqrt{6}} \left[\frac{J}{C_0 \sqrt{2x}}\right]^{1/3}\right) = .01 \Rightarrow \frac{h_{99}}{\sqrt{6}} \left[\frac{J}{C_0 \sqrt{2x}}\right]^{1/3} = 2.2924$$

$$h_{99} = 5.6152 \left[\frac{C_0 \sqrt{2x}}{J}\right]^{1/3}$$

jet width  $\propto x^{1/3}$  as  $v$  increase  
 width, whereas  $J$  decreases width

$$Re_x = \frac{x u_0(x)}{\nu} = \left(\frac{3Jx}{4\sqrt{6}C_0}\right)^{1/3} \quad Re_{hm} = 5.6152 \left(\frac{3Jx}{4\sqrt{6}C_0}\right)^{1/3}$$

Stability:  $Re \gg 1$  due inflection points.