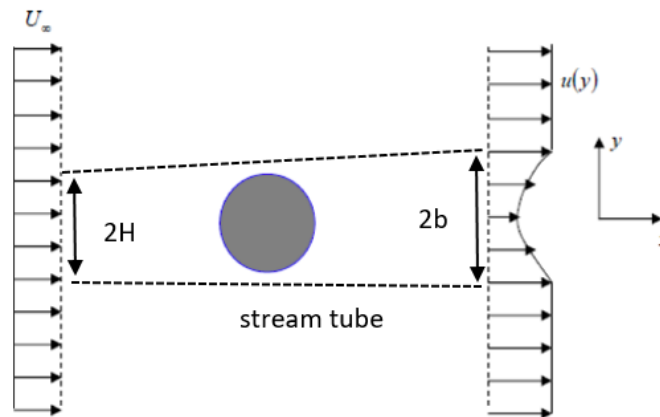


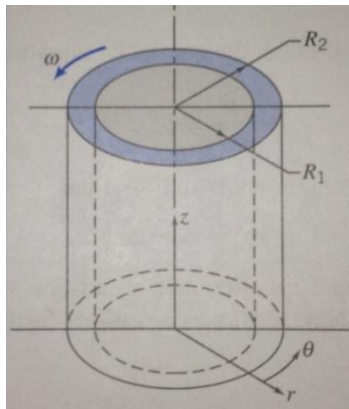
**The exam is closed book and closed notes.**

1. Consider an experiment in which the drag on a two-dimensional body immersed in a steady incompressible flow can be determined from measurement of velocity distribution far upstream and downstream of the body as shown in Figure below. Velocity far upstream is the uniform flow  $U_\infty$ , and that in the wake of the body is measured to be  $u(y) = \frac{U_\infty}{2} \left( \frac{y^2}{b^2} + 1 \right)$ , which is less than  $U_\infty$  due to the drag of the body. Assume that there is a stream tube with inlet height of  $2H$  and outlet height of  $2b$  as shown in Figure below. (a) Determine the relationship between  $H$  and  $b$  using the continuity equation. (b) Find the drag per unit length of the body as a function of  $U_\infty$ ,  $b$  and  $\rho$ .

(Hint: Momentum Equation  $\sum F_x = \int u\rho(\underline{V} \cdot \underline{n})dA$ )



2. A Newtonian fluid of density  $\rho$  and viscosity  $\mu$  fills the annular gap between concentric cylinders as shown below. The inner cylinder of radius  $R_1$  is stationary, and the outer cylinder of radius  $R_2$  rotates steadily at a constant angular speed  $\omega$ . The flow between the cylinders is laminar and incompressible. Assume steady, purely circulating motion, 2D flow, circumferentially symmetric pressure, and neglect gravity. Simplify the continuity and momentum equations and use appropriate boundary conditions to determine the velocity distribution in the gap.



The equations of motion of an incompressible Newtonian fluid with constant density and viscosity in cylindrical coordinates  $(r, \theta, z)$  with velocity components  $(v_r, v_\theta, v_z)$ :

Continuity:

$$\frac{1}{r} \frac{\partial}{\partial r} (r v_r) + \frac{1}{r} \frac{\partial}{\partial \theta} (v_\theta) + \frac{\partial}{\partial z} (v_z) = 0$$

r-momentum:

$$\rho \left( \frac{\partial v_r}{\partial t} + v_r \frac{\partial v_r}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_r}{\partial \theta} + v_z \frac{\partial v_r}{\partial z} - \frac{v_\theta^2}{r} \right) = \rho g_r - \frac{\partial p}{\partial r} + \mu \left[ \frac{\partial}{\partial r} \left( \frac{1}{r} \frac{\partial}{\partial r} (r v_r) \right) + \frac{1}{r^2} \frac{\partial^2 v_r}{\partial \theta^2} + \frac{\partial^2 v_r}{\partial z^2} - \frac{2}{r^2} \frac{\partial v_\theta}{\partial \theta} \right]$$

$\theta$ -momentum:

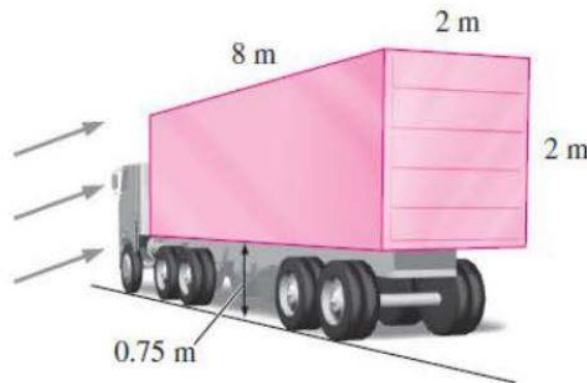
$$\rho \left( \frac{\partial v_\theta}{\partial t} + v_r \frac{\partial v_\theta}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_\theta}{\partial \theta} + v_z \frac{\partial v_\theta}{\partial z} + \frac{v_r v_\theta}{r} \right) = \rho g_\theta - \frac{1}{r} \frac{\partial p}{\partial \theta} + \mu \left[ \frac{\partial}{\partial r} \left( \frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) \right) + \frac{1}{r^2} \frac{\partial^2 v_\theta}{\partial \theta^2} + \frac{\partial^2 v_\theta}{\partial z^2} + \frac{2}{r^2} \frac{\partial v_r}{\partial \theta} \right]$$

z-momentum:

$$\rho \left( \frac{\partial v_z}{\partial t} + v_r \frac{\partial v_z}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_z}{\partial \theta} + v_z \frac{\partial v_z}{\partial z} \right) = \rho g_z - \frac{\partial p}{\partial z} + \mu \left[ \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial v_z}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 v_z}{\partial \theta^2} + \frac{\partial^2 v_z}{\partial z^2} \right]$$

3. During major windstorms, high vehicles such as RVs and semis may be thrown off the road, especially when they are empty and in open areas. Consider a 5000-kg semi that is 8m long, 2m high, and 2m wide. The distance between the bottom of the truck and the road is 0.75m. The truck is exposed to winds from its side surface. (a) Determine the wind velocity that will tip the truck over to its side. Take the air density to be  $1.1 \text{ kg/m}^3$  and assume the weight to be uniformly distributed. (b) If the wind arrives with a  $45^\circ$  angle on the side of the truck, does the tip velocity increase or decrease (suppose same  $C_D$ )? Why? (c) If you are to repeat the experiment for a 2 times smaller model using Reynolds similarity and the new semi weights 2000-kg, determine the new tipping velocity (consider that the distance between the bottom of the truck and the road is 0.5m).

**Hint:**  $C_D = \frac{D}{0.5\rho AV^2}$  (use square rod)      $Re_{full\ scale} = \frac{\rho U(2m)}{\mu}$

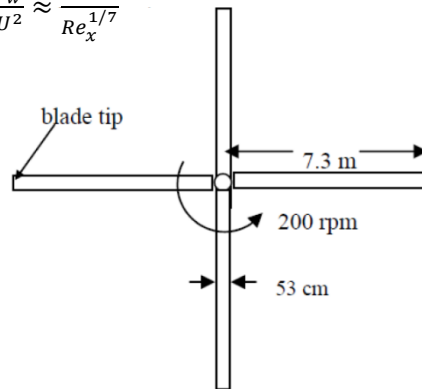


<p><b>Square rod</b></p> <p>Sharp corners: <math>C_D = 2.2</math></p> <p>Round corners (<math>r/D = 0.2</math>): <math>C_D = 1.2</math></p>	<p><b>Rectangular rod</b></p> <p>Sharp corners:</p> <p>Round front edge:</p>	<table border="1"> <thead> <tr> <th><math>L/D</math></th> <th><math>C_D</math></th> </tr> </thead> <tbody> <tr> <td>0.0*</td> <td>1.9</td> </tr> <tr> <td>0.1</td> <td>1.9</td> </tr> <tr> <td>0.5</td> <td>2.5</td> </tr> <tr> <td>1.0</td> <td>2.2</td> </tr> <tr> <td>2.0</td> <td>1.7</td> </tr> <tr> <td>3.0</td> <td>1.3</td> </tr> </tbody> </table> <p>* Corresponds to thin plate</p> <table border="1"> <thead> <tr> <th><math>L/D</math></th> <th><math>C_D</math></th> </tr> </thead> <tbody> <tr> <td>0.5</td> <td>1.2</td> </tr> <tr> <td>1.0</td> <td>0.9</td> </tr> <tr> <td>2.0</td> <td>0.7</td> </tr> <tr> <td>4.0</td> <td>0.7</td> </tr> </tbody> </table>	$L/D$	$C_D$	0.0*	1.9	0.1	1.9	0.5	2.5	1.0	2.2	2.0	1.7	3.0	1.3	$L/D$	$C_D$	0.5	1.2	1.0	0.9	2.0	0.7	4.0	0.7
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<p><b>Circular rod (cylinder)</b></p> <p>Laminar: <math>C_D = 1.2</math></p> <p>Turbulent: <math>C_D = 0.3</math></p>	<p><b>Elliptical rod</b></p>	<table border="1"> <thead> <tr> <th rowspan="2"><math>L/D</math></th> <th colspan="2"><math>C_D</math></th> </tr> <tr> <th>Laminar</th> <th>Turbulent</th> </tr> </thead> <tbody> <tr> <td>2</td> <td>0.60</td> <td>0.20</td> </tr> <tr> <td>4</td> <td>0.35</td> <td>0.15</td> </tr> <tr> <td>8</td> <td>0.25</td> <td>0.10</td> </tr> </tbody> </table>	$L/D$	$C_D$		Laminar	Turbulent	2	0.60	0.20	4	0.35	0.15	8	0.25	0.10										
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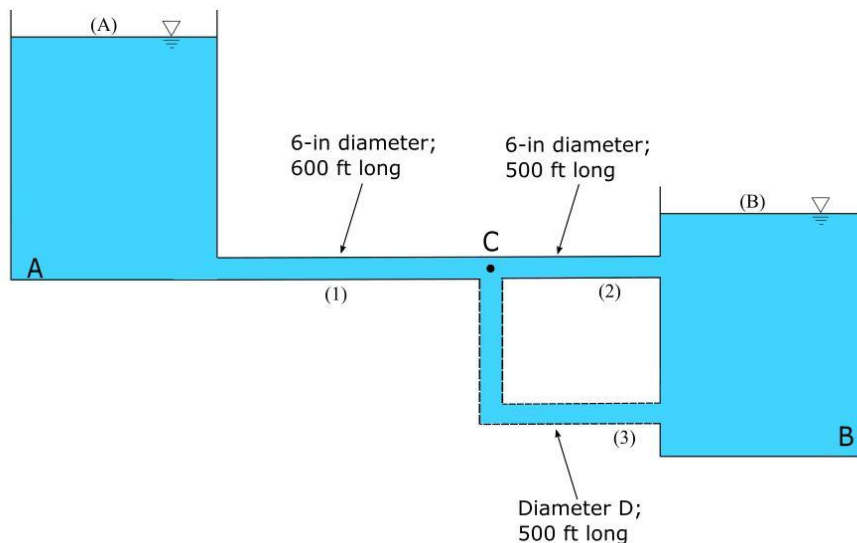
4. A helicopter rotor rotates at  $\omega = 20.94 \text{ rad/s}$  in air ( $\rho = 1.2 \text{ kg/m}^3$  and  $\mu = 1.8 \times 10^{-5} \text{ kg/m-s}$ ). Each blade has a chord length of 53 cm and extends a distance of 7.3 m from the center of the rotor hub. Assume that the blades can be modeled as very thin flat plates at a zero angle of attack. (a) At what radial distance from the hub center is the flow turbulent ( $Re_{crit} = 5 \times 10^5$ ). (b) Find the boundary layer thickness at the blade tip trailing edge (c) At what rotor angular velocity does the wall shear stress at the blade tip trailing edge become  $80 \text{ N/m}^2$ ?

Hint:  $Re = \frac{\rho U C}{\mu}$  where  $U = r\omega$ ,  $C$ : Chord Length

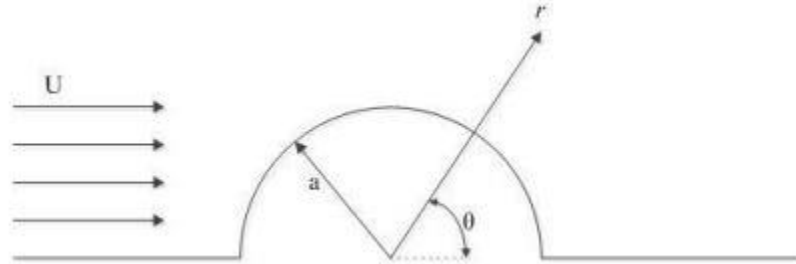
$$\text{Turbulent BL: } \frac{\delta}{x} \approx \frac{0.16}{Re^{1/7}}, \quad c_f = \frac{2\tau_w}{\rho U^2} \approx \frac{0.027}{Re_x^{1/7}}$$



5. The flowrate between tank A and tank B shown in the figure is to be increased by 30% (i.e. from  $Q$  to  $1.30Q$ ) by the addition of a second pipe (indicated by the dotted lines) running from node C to tank B. If the elevation of the free surface in tank A is 25 ft above that in tank B, determine the diameter,  $D$ , of this new pipe. Neglect minor losses and assume that the friction factor for each pipe is 0.02.



6. A practice facility for the Cal football team is to be covered by a semi-circular cylindrical dome, as shown in the sketch. The dome with radius  $a$  is placed normal to the cross flow of velocity  $U$ . Assume that the dome is very long (into the page) so you can ignore end effects.



- a. The stream function and velocity potential for this flow  $0 \leq \theta \leq \pi$  are given by

$$\Psi = Ur \left( 1 - \frac{a^2}{r^2} \right) \sin \theta$$

$$\Phi = Ur \left( 1 + \frac{a^2}{r^2} \right) \cos \theta$$

Use potential flow theory to find the tangential velocity along the surface of the dome ( $r = a$ ) as a function of the angle  $\theta$ .

Hint:  $u_r = \frac{\partial \Phi}{\partial r}$ ,  $u_\theta = \frac{1}{r} \frac{\partial \Phi}{\partial \theta}$

- b. Choose a reference condition to be  $p = p_0$  at  $z=0$  (on the ground) far from the cylindrical dome. Use the Bernoulli equation to find the pressure field on the surface of the dome ( $r = a$ ) as a function of  $\theta$ . Do NOT neglect hydrostatic variations.
- c. If the vertical force on the dome per unit length (upwards) is given by  $F_{lift} = \frac{5}{3} \rho U^2 a$ , find the mass per unit length ( $m_c$ ) required to keep the cylindrical dome on the ground when the wind blows at 30m/s, with air density  $\rho=1.2\text{kg/m}^3$ , and  $a=50\text{m}$ .

**1. Solution**

a) Continuity:

$$2\rho HU_\infty = \rho \int_{-b}^b u(y) dy = \rho \int_{-b}^b \frac{U_\infty}{2} \left( \frac{y^2}{b^2} + 1 \right) dy \quad (2)$$

$$2\rho HU_\infty = \rho \frac{U_\infty}{2} \int_{-b}^b \left( \frac{y^2}{b^2} + 1 \right) dy = \rho \frac{U_\infty}{2} \left( \frac{y^3}{3b^2} + y \right) \Big|_{-b}^b \quad (1)$$

$$2H = \frac{1}{2} \left( \frac{b^3}{3b^2} + b + \frac{b^3}{3b^2} + b \right) = \frac{1}{2} \left( \frac{8}{3} b \right) = \frac{4}{3} b \quad (1)$$

$$H = \frac{2b}{3} \quad (1)$$

b) x-momentum:

$$\Sigma F_x = \int u\rho(\underline{V} \cdot \underline{n}) dA \quad (2)$$

Drag per unit length:

$$-F_D = -\rho U_\infty^2 (2H) + \rho \int_{-b}^b u^2(y) dy \quad (1)$$

$$F_D = \rho U_\infty^2 (2H) - \rho \int_{-b}^b \left[ \frac{U_\infty}{2} \left( \frac{y^2}{b^2} + 1 \right) \right]^2 dy = \rho U_\infty^2 (2H) - \rho \frac{U_\infty^2}{4} \int_{-b}^b \left( \frac{y^2}{b^2} + 1 \right)^2 dy$$

Calculating integral:

$$\int_{-b}^b \left( \frac{y^2}{b^2} + 1 \right)^2 dy = \int_{-b}^b \left( \frac{y^4}{b^4} + \frac{2y^2}{b^2} + 1 \right) dy = \left[ \frac{y^5}{5b^4} + \frac{2y^3}{3b^2} + y \right]_{-b}^b = 2 \left( \frac{b}{5} + \frac{2b}{3} + b \right) = \frac{56}{15} b \quad (1)$$

Entering into the momentum equation:

$$F_D = 2\rho HU_\infty^2 - \frac{1}{4} \rho U_\infty^2 \left( \frac{56}{15} b \right) = \rho U_\infty^2 \left( \frac{4}{3} b - \frac{14}{15} b \right) = \rho U_\infty^2 \frac{2b}{5} \quad (1)$$

**2. Solution**

ASSUMPTIONS:

1. Steady flow ( $\frac{\partial}{\partial t} = 0$ )
2. Purely circumferential flow ( $v_r = v_z = 0$ )
3. 2D flow ( $\frac{\partial}{\partial z} = 0$ )
4. circumferentially symmetric pressure ( $\frac{\partial p}{\partial \theta} = 0$ )
5. Negligible gravity

ANALYSIS:

Continuity:

$$\frac{1}{r} \frac{\partial}{\partial r} (rv_r) + \frac{1}{r} \frac{\partial}{\partial \theta} (v_\theta) + \frac{\partial}{\partial z} (v_z) = 0$$

$$0(2) + \frac{1}{r} \frac{\partial}{\partial \theta} (v_\theta) + 0(2) = 0$$

$$\frac{\partial v_\theta}{\partial \theta} = 0 \quad (6) \quad (2)$$

$\theta$ -momentum:

$$\begin{aligned} & \rho \left( \frac{\partial v_\theta}{\partial t} + v_r \frac{\partial v_\theta}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_\theta}{\partial \theta} + v_z \frac{\partial v_\theta}{\partial z} + \frac{v_r v_\theta}{r} \right) \\ & = \rho g_\theta - \frac{1}{r} \frac{\partial p}{\partial \theta} + \mu \left[ \frac{\partial}{\partial r} \left( \frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) \right) + \frac{1}{r^2} \frac{\partial^2 v_\theta}{\partial \theta^2} + \frac{\partial^2 v_\theta}{\partial z^2} + \frac{2}{r^2} \frac{\partial v_r}{\partial \theta} \right] \\ \rho(0(1) + 0(2) + 0(6) + 0(2,3) + 0(2)) & = 0(5) - 0(4) + \mu \left[ \frac{\partial}{\partial r} \left( \frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) \right) + 0(6) + 0(3) + 0(2) \right] \end{aligned}$$

$$\frac{\partial}{\partial r} \left( \frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) \right) = 0 \quad (2)$$

Integrate:

$$\frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) = C_1$$

$$\frac{\partial}{\partial r} (r v_\theta) = C_1 r \quad (1)$$

Integrate again:

$$r v_\theta = C_1 \frac{r^2}{2} + C_2$$

$$v_\theta = C_1 \frac{r}{2} + \frac{C_2}{r} \quad (1)$$

B.C.:

$$1) \text{ At } r = R_1, v_\theta = 0 \quad (0.75)$$

$$2) \text{ At } r = R_2, v_\theta = R_2 \omega \quad (0.75)$$

Replace:

$$0 = C_1 \frac{R_1}{2} + \frac{C_2}{R_1}$$

$$R_2 \omega = C_1 \frac{R_2}{2} + \frac{C_2}{R_2}$$

And find:

$$C_1 = \frac{2\omega}{1 - \left(\frac{R_1}{R_2}\right)^2} \quad (0.25)$$

$$C_2 = \frac{-\omega R_1^2}{1 - \left(\frac{R_1}{R_2}\right)^2} \quad (0.25)$$

Substitute into the expression for  $v_\theta$ :

$$v_{\theta} = \frac{\omega R_1}{1 - \left(\frac{R_1}{R_2}\right)^2} \left(\frac{r}{R_1} - \frac{R_1}{r}\right) \quad (1)$$

### 3. Solution

(a)

From table:  $L/D=2/2 \rightarrow C_D = 2.2$  (2)

When the truck is first tipped, the wheels on the wind-loaded side of the truck will be off the ground, and thus all reaction forces from the ground will act on wheels on the other side. Taking the moment about an axis passing through these wheels and setting it equal to zero gives the required drag force and consequently the wind speed:

$$\sum M = 0 \rightarrow M_{drag} - M_{weight} = 0 \quad (1)$$

$$M_{drag} = F_D \times arm_{drag} \quad (1)$$

$$F_D = 0.5\rho AV^2 C_D = 0.5(1.1)(8)(2)V^2(2.2) = 19.36V^2 \quad (1)$$

$$arm_{drag} = \frac{2}{2} + 0.75 = 1.75m$$

$$M_{weight} = mg \times arm_{weight} \quad (1)$$

$$arm_{weight} = \frac{2}{2} = 1$$

$$\rightarrow 19.36V^2(1.75) - 49050(1) = 0$$

$$\rightarrow V = 38.05m/s \quad (1)$$

(b)

If the wind arrives with a  $45^\circ$  angle, the normal area decreases, reducing  $F_D$  and increasing the tipping velocity. (1)

(c)

Reynolds similarity:

$$\frac{\rho VL}{\mu} = \frac{\rho V_m L_m}{\mu} \quad (1)$$

$$V_m = \frac{V L}{L_m} = \frac{2V}{1} = 2V$$

$$M_{drag} = F_D \times arm_{drag}$$

$$F_D = 0.5\rho AV^2 C_D = 0.5(1.1)(4)(1)(2V)^2(2.2) = 19.36V^2$$



$$arm_{drag} = \frac{1}{2} + 0.5 = 1m$$

$$M_{weight} = mg \times arm_{weight}$$

$$arm_{weight} = \frac{1}{2} = 0.5$$

$$\rightarrow 19.36V^2(1) - 19620(0.5) = 0$$

$$\rightarrow V = 22.51m/s \quad (1)$$

#### 4. Solution

a)

$$U = r\omega$$

$$Re = \frac{\rho U c}{\mu} = \frac{\rho r \omega c}{\mu} \quad (1)$$

$$Re_{crit} = \frac{\rho r_{crit} \omega c}{\mu}$$

$$r_{crit} = \frac{Re_{crit} \mu}{\rho \omega c} = \frac{(5E5)(1.8E-5)}{(1.2)(20.94)(0.53)} = 0.68 m \quad (1)$$

b) At the tip trailing edge:

$$Re = \frac{\rho r_{tip} \omega c}{\mu} = \frac{(1.2)(7.3)(20.94)(0.53)}{(1.8E-5)} = 5,401,124 \quad (1)$$

$$\frac{\delta}{x} \approx \frac{0.16}{Re^{1/7}} \quad (1)$$

$$\delta \approx \frac{0.16(0.53)}{(5401124)^{1/7}} = 0.00926 m = 9.26 mm \quad (2)$$

c)

$$\tau_w = 0.0135 \frac{\rho U^2}{Re_x^{1/7}} = \frac{0.0135 \mu^{1/7} \rho^{6/7} U^{13/7}}{x^{1/7}} \quad (2)$$

Re-arrange to find  $U$ :

$$U = r_{tip}\omega = \left( \frac{\tau_w x^{1/7}}{0.0135\mu^{1/7}\rho^{6/7}} \right)^{7/13} \quad (1)$$

$$\omega = \frac{1}{(7.3)} \left( \frac{(80.0)(0.53)^{1/7}}{0.0135 (1.8E - 5)^{1/7}(1.2)^{6/7}} \right)^{7/13} = 29.88 \text{ rad/s} \quad (1)$$

## 5. Solution

With the single pipe:

$$\left( \frac{p}{\rho g} + \frac{\alpha V^2}{2g} + z \right)_A = \left( \frac{p}{\rho g} + \frac{\alpha V^2}{2g} + z \right)_B + h_f \quad (2)$$

Where  $p_a = p_b = 0$ ,  $V_a = V_b = 0$ ,  $z_a = 25ft$ ,  $z_b = 0$ .

Thus,

$$z_a = f \frac{L_1 + L_2}{D_1} \frac{V_1^2}{2g} \quad (1)$$

$$25ft = 0.02 \frac{(600 + 500)}{6/12} \frac{V_1^2}{2(32.2)}$$

$$V_1 = 5.96 \text{ ft/s}$$

$$Q_1 = A_1 V_1 = \frac{\pi}{4} (0.5^2) 5.96 = 1.188 \text{ ft}^3/\text{s} \quad (1)$$

With the second pipe  $Q=1.3Q=1.54 \text{ ft}^3/\text{s}$ .

Therefore,  $Q_1 = Q_2 + Q_3$  and

$$V_1 = \frac{Q_1}{A_1} = \frac{1.52}{\frac{\pi}{4} (0.5^2)} = 7.84 \text{ ft/s} \quad (2)$$

For fluid flowing from A to B through pipes 1 and 2:

$$z_a = h_{L1} + h_{L2} = f \frac{L_1}{D_1} \frac{V_1^2}{2g} + f \frac{L_2}{D_2} \frac{V_2^2}{2g} \quad (1)$$

$$z_a = 0.02 \frac{600}{0.5} \frac{7.84^2}{2(32.2)} + 0.02 \frac{500}{0.5} \frac{V_2^2}{2(32.2)}$$

Hence,  $V_2 = 2.6 \text{ ft/s}$

And

$$Q_2 = A_2 V_2 = \frac{\pi}{4} (0.5^2) 2.60 = 0.511 ft^3/s$$

Thus,

$$Q_3 = Q_1 - Q_2 = 1.03 ft^3/s \quad (1)$$

For fluid flowing from A to B through pipes 1 and 3,

$$z_a = h_{L1} + h_{L3} = f \frac{L_1 V_1^2}{D_1 2g} + f \frac{L_3 V_3^2}{D_3 2g}$$

Where

$$V_3 = \frac{Q_3}{A_3} = \frac{1.03}{\frac{\pi}{4} (D_3^2)} = \frac{1.31}{D_3^2}$$

$$25 ft = 0.02 \frac{(600)}{0.5} \frac{7.84^2}{2(32.2)} + 0.02 \frac{(500)}{D_3} \frac{\left(\frac{1.31}{D_3^2}\right)^2}{2(32.2)}$$

$$D_3 = 0.662 ft \quad (1)$$

## 6. Solution

- a. Find the tangential velocity along the surface of the dome

$$u_\theta = \frac{1}{r} \frac{\partial \Phi}{\partial \theta} = \frac{1}{r} \frac{\partial}{\partial \theta} \left( Ur \left( 1 + \frac{a^2}{r^2} \right) \cos \theta \right) \quad (2)$$

$$u_\theta = \frac{1}{r} \left( -Ur \left( 1 + \frac{a^2}{r^2} \right) \sin \theta \right)$$

At  $r=a$ , we have

$$u_\theta = -2U \sin \theta \quad (1)$$

- b. Apply Bernoulli between any 2 points

$$p_0 + \frac{1}{2} \rho V_0^2 + \gamma z_0 = p_s + \frac{1}{2} \rho V_s^2 + \gamma z_s \quad (3)$$

s in on the surface.

$$p_0 + \frac{1}{2} \rho U^2 + 0 = p_s + \frac{1}{2} \rho (-2U \sin \theta)^2 + \gamma a \sin \theta$$

$$p_s = p_0 + \frac{1}{2} \rho U^2 - 2\rho (U \sin \theta)^2 - \gamma a \sin \theta \quad (2)$$

- c.

---

$$\sum F_z = 0 \quad (1)$$

$$F_{lift} - F_w = 0$$

$$\frac{5}{3}\rho U^2 a = m_c g$$

$$m_c = \frac{5}{3}(1.2)(30)^2(50)\left(\frac{1}{9.8}\right)$$

$$m_c = 9170 \text{ kg/m} \quad (1)$$