57:020 Mechanics of Fluids and Transport Processes

Simulation of Turbulent Flow around an Airfoil

Submitted to: Professor Frederick Stern

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University ID:	
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Department:	
Group:	

Date: Month/Date/2016

I. CFD PreLab 2 Questions (submitted before PreLab1, 10 points)

II. Test and Simulation Design (10 points)

Using your own words to describe the purpose of the CFD simulation.

III. CFD Process (10 points)

Using either table or Flow chart to describe the correct CFD Process, which options will be used in CFD PreLab 1 and CFD Lab 1, and what parameters (with values) need to be specified for each option.

IV. Data Analysis and Discussion (50 points)

1. Figures, data and questions in CFD PreLab 2 (Simulation of Turbulent flow around an airfoil, 20 points)

A. Validation using EFD Lab 3 data using k-e model (7 points)

A. van	A. Validation using ETD Lab 3 data using k-e model (7 points)				
	EFD	CFD	Relative Error (%)		
Lift coef.					
Drag coef.					
	EFD-2D-	CFD	Relative		
	Benchmark		Error (%)		
Lift coef.	0.444		` /		
Drag coef.	0.025				
	•			Figure 1. Residuals	
Figure 2. Pressure coefficient distribution (CFD and EFD)			distribution	Figure 3. Contour of pressure	
Figure 4	. Contour of	velocity	magnitude	Figure 5. Velocity vectors	

Figure 6. Streamlines close to airfoil surface	
B. Inviscid Flow Simulation (7 points)	
Lift Coefficient (CFD)=	
Drag Coefficient (CFD)=	
	Figure 1. Residuals
Figure 2. Pressure coefficient distribution (CFD only)	Figure 3. Contour of pressure
Figure 4. Contour of velocity magnitude	Figure 5. Velocity vectors
Figure 6. Streamlines close to airfoil surface	

C. Questions in CFD PreLab 2 (6 points)

(1). Does inviscid flow has boundary layer near the wall? Zoom in the near wall region and
describe the differences of velocity vectors near the airfoil surface for inviscid and
viscous flows.

Answer:

- (2). What are the correct boundary conditions for velocity and pressure at "inlet" and "outlet". Answer:
- (3). What are the values for lift and drag coefficients for inviscid flow around the airfoil? Are they both zero?

Answer:

- (4). Where are the highest and lowest locations for pressure and velocity magnitude? Why? Is pressure constant for inviscid flows around airfoil?

 Answer:
- (5). For turbulent flow around airfoil, try to qualitatively explain why there is a lift force (vertical up) on an airfoil using the contour plot of pressure or the plot of pressure coefficient distribution.

Answer:

2. Figures, data, and questions in CFD Lab 2 (Parametric Studies of turbulent flow around an airfoil, 30 points)

A. Figures and tables in CFD Lab 2 (Effect of Angle of attack, 6 points).

71. 1150	nes ana tac	7105 111	CI B Euc 2	(Liteet of Fingle of attack, 6 points).
	EFD	CFD	Relative	
	LID	CID		
			error (%)	
Lift coef.				
Drag coef.				
	EED AD	CED	D -1-42	
	EFD-2D-	CFD	Relative	
	Benchmark		Error (%)	
Lift coef.	1.439			
Drag coef.	0.171			Figure 1 Deciduels
		l.		Figure 1. Residuals
Eigene 2	Figure 2. Pressure coefficient distribution		ما المحملات الم	
Figure 2.			distribution	
	(CFD and	d EFD)		Figure 2 Contour of programs
(612 4114 212)			Figure 3. Contour of pressure	

Figure 4. Velocity vectors	Figure 5. Streamlines close to airfoil surface

B. Figures and tables in CFD Lab 2 (Effect of turbulent models, k-w, 6 points).

B. Figi	ares and ta	bles in	CFD Lab 2 (E	effect of turbulent models, k-w, 6 points).
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	EFD	CFD	Relative error (%)	
Lift coef.			error (76)	
Drag coef.				
		•		
	EFD-2D-	CFD	Relative	
	Benchmark		Error (%)	
Lift coef.	1.439			Fi 4 5
Drag coef.	0.1710.171			Figure 1. Residuals
Figure 2	Pressure co	efficient	distribution	
riguic 2.	(CFD an		distribution	A G
	(CI D an	u Li D)		Figure 3. Contour of pressure
_	. 4 37 1	•,		
F	igure 4. Vel	ocity ve	ctors	Figure 5. Streamlines close to airfoil surface

#### C. Questions in CFD Lab 2 (18 points)

C.1. For effect of angle of attack:

(1). Which angle of attack simulation requires more iterations to converge? Answer: .....

- (2). Which angle of attack produces higher lift/drag coefficients? Why? Answer: .....
- (3). Describe the differences of streamline distributions near the trailing edge of airfoil surface for these two different angles of attack. Do you observe separations for both? If so, does the separation occur at the same location?

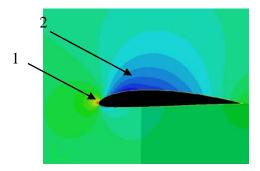
  Answer: .....

#### C.2. For effect of different turbulence models:

- (1). Do the two different turbulence models have the same convergence path? If not, which one requires more iterations to converge.

  Answer: .....
- (2). Do the two different turbulence models predict the same results? If not, which model predicts more accurately by comparing with EFD data?

  Answer: .....
- **C.3.** For the following contour plot, qualitatively compare the values of pressure and velocity magnitude at point 1 and 2, if the flow is from left to right. Which location has higher pressure and which location has higher velocity magnitude? Why? Answer: .....



#### V. Conclusions (15 points)

A. Conclusions regarding achieving purpose of simulation (3 points)

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B. What I have learned from CFD Lab. (6 points)

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C. Comments on the hands-on experience, the software interface, and overall lab performance (4 points)

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D. Suggestions and improvements (2 points)