

Review for Exam 2

Chapter 5: Mass, Bernoulli, and Energy equation

- Flow rate and conservation of mass
general case



$$\begin{aligned}
 Q &= \int_{CS} \underline{V} \cdot \underline{n} dA \\
 &= \int_{CS} |\underline{V}| \cos \theta dA \\
 \dot{m} &= \int_{CS} \rho (\underline{V} \cdot \underline{n}) dA
 \end{aligned}$$

average velocity: $\bar{V} = \frac{Q}{A}$

General Form of Continuity Equation

$$\frac{dM}{dt} = 0 = \frac{d}{dt} \int_{CV} \rho dV + \int_{CS} \rho \underline{V} \cdot \underline{dA}$$

or

$$\underbrace{\int_{CS} \rho \underline{V} \cdot \underline{dA}} = \underbrace{-\frac{d}{dt} \int_{CV} \rho dV}$$

net rate of outflow
of mass across CS

rate of decrease of
mass within CV

Simplification of Continuity Equation

1. Steady flow: $-\frac{d}{dt} \int_{CV} \rho dV = 0$

2. $\underline{V} = \text{constant}$ over discrete \underline{dA} (flow sections):

$$\int_{CS} \rho \underline{V} \cdot \underline{dA} = \sum_{CS} \rho \underline{V} \cdot \underline{A}$$

3. Incompressible fluid ($\rho = \text{constant}$)

$$\int_{CS} \underline{V} \cdot \underline{dA} = -\frac{d}{dt} \int_{CV} dV \quad \text{conservation of volume}$$

4. Steady One-Dimensional Flow in a Conduit:

$$\sum_{CS} \rho \underline{V} \cdot \underline{A} = 0$$

$$-\rho_1 V_1 A_1 + \rho_2 V_2 A_2 = 0$$

$$\text{for } \rho = \text{constant} \quad Q_1 = Q_2$$

- **Bernoulli Equations, Application, and Limitation**

Assume the flow is

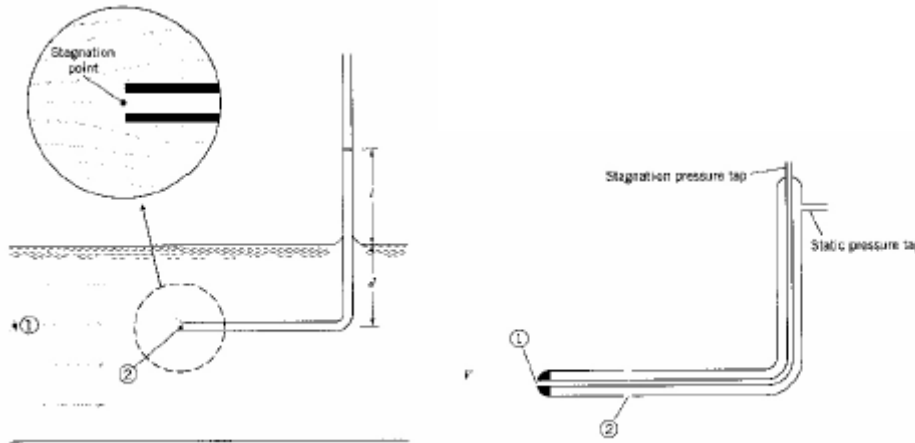
- Irrotational
- Inviscid
- Incompressible

$$p_1 + \frac{1}{2}\rho V_1^2 + \gamma z_1 = p_2 + \frac{1}{2}\rho V_2^2 + \gamma z_2$$

Bernoulli equation for steady state

Application:

Stagnation tube (right), Pitot tube (left)



• **Energy Equation**

$$\dot{Q} - \dot{W} = \frac{d}{dt} \int_{CV} \rho \left(\frac{V^2}{2} + gz + u \right) dV + \int_{CS} \rho \left(\frac{V^2}{2} + gz + u \right) \underline{V} \cdot \underline{dA}$$

rate of work done by system
rate of change of energy in CV
flux of energy out of CV (ie, across CS)

rate of heat transfer to system

• **Simplified form of energy equation for steady one dimensional incompressible flow**

Simplification:

- No acceleration normal to the stream lines, pressure is hydrostatically distributed.
- Internal energy u is considered as constant.
- Shaft work defined as $\dot{W}_s = \dot{W}_t - \dot{W}_p$

• Head loss defined as
$$h_L = \frac{u_2 - u_1}{g} - \frac{\dot{Q}}{g\dot{m}}$$

The final form of simplified energy equation:

$$\frac{p_1}{\gamma} + \alpha_1 \frac{V_1^2}{2g} + z_1 + h_p = \frac{p_2}{\gamma} + \alpha_2 \frac{V_2^2}{2g} + z_2 + h_t + h_L$$

$$h_p = \dot{W}_p / \dot{m}g$$

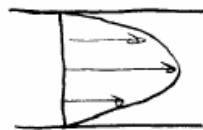
$$h_t = \dot{W}_t / \dot{m}g$$

$$\alpha = \frac{1}{A\bar{V}^2} \int_A V^3 dA = \text{kinetic energy correction factor}$$

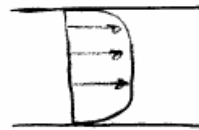
$$\bar{V} = \frac{1}{A} \int_A V dA = Q/A$$

V_1 & V_2 are average velocities

note that: $\alpha = 1$ if V is constant across the flow section
 $\alpha > 1$ if V is nonuniform



laminar flow $\alpha = 2$



turbulent flow $\alpha = 1.05 \sim 1$ may be used

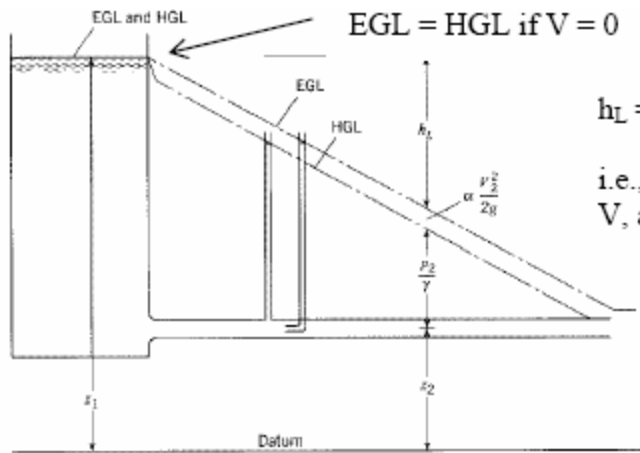
• **HGL and EGL**

$$\frac{p_1}{\gamma} + \alpha_1 \frac{V_1^2}{2g} + z_1 + h_p = \frac{p_2}{\gamma} + \alpha_2 \frac{V_2^2}{2g} + z_2 + h_t + h_L$$

Define	$HGL = \frac{p}{\gamma} + z$	} point-by-point application is graphically displayed
	$EGL = \frac{p}{\gamma} + z + \alpha \frac{V^2}{2g}$	

HGL corresponds to pressure tap measurement + z

EGL corresponds to stagnation tube measurement + z



$$h_L = f \frac{L}{D} \frac{V^2}{2g}$$

i.e., linear variation in L for D, V, and f constant

f = friction factor
f = f(Re)

Chapter 6: Momentum Analysis of Flow System

• Momentum Equation

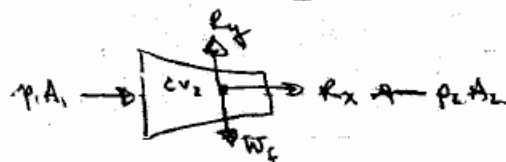
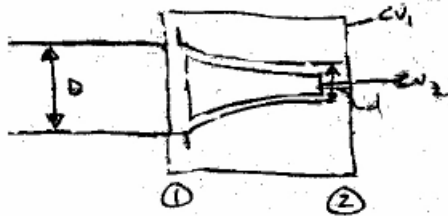
$$\sum \mathbf{F} = \frac{d}{dt} \int_{CV} \mathbf{V} \rho d\forall + \int_{CS} \mathbf{V} \rho \mathbf{V}_R \cdot d\mathbf{A}$$

sum of all external forces acting on CV, i.e., pressure force, forces transmitted through solids, etc.

relative velocity = $\mathbf{V} - \mathbf{V}_{CS}$

Special form for steady flow with uniform flow across discrete CS

$$\sum \mathbf{F} = \sum \rho \mathbf{V} (\mathbf{V} \cdot \mathbf{A})$$



(R_x, R_y) = reaction force on fluid



(R_x, R_y) = reaction force on nozzle

Important features for momentum equation:

1. Vector equations
2. Carefully define CV to include all external body and surface faces
3. Velocity must be referenced to a non-accelerating inertial frame.
4. Steady vs Unsteady flow
5. Uniform vs Nonuniform flow
6. Always use gage pressure
7. Pressure condition at a jet exit

Chapter 7: Dimensional Analysis and Modeling

- Buckingham Π theorem

$$F(A_1, \dots, A_n) = 0$$

$$\Rightarrow f\left(\pi_1, \dots, \pi_{\frac{n-\hat{m}}{r}}\right) = 0 \quad \hat{m} = m \text{ usually}$$

significant reduction in number of variables which reduces number of experiments or calculations required

- Methods for determining Π_i 's

1. Functional Relationship Method

Identify functional relationships $F(A_i)$ and $f(\Pi_j)$ by first determining A_i 's and then evaluating Π_j 's

- | | |
|------------------------|-----------|
| a. Inspection | intuition |
| b. Step-by-step Method | text |
| c. Exponent Method | class |

2. Nondimensionalize governing differential equations and initial and boundary conditions

- Common Dimensionless Parameters for Fluid Flow Problems

Re, Fr, We, Ma, C_p, etc

- Nondimensionalization of the Basic Equation

$$\nabla \cdot \mathbf{V} = 0$$

$$\frac{D\mathbf{V}}{Dt} = -\nabla p + Re^{-1} \nabla^2 \mathbf{V}$$

$$\mathbf{V} = \mathbf{V}/U_0$$

$$\mathbf{X} = \mathbf{X}/L$$

$$t = tU_0/L$$

$$p = (p + \rho gz) / \rho U_0^2$$

- Similarity and Model testing

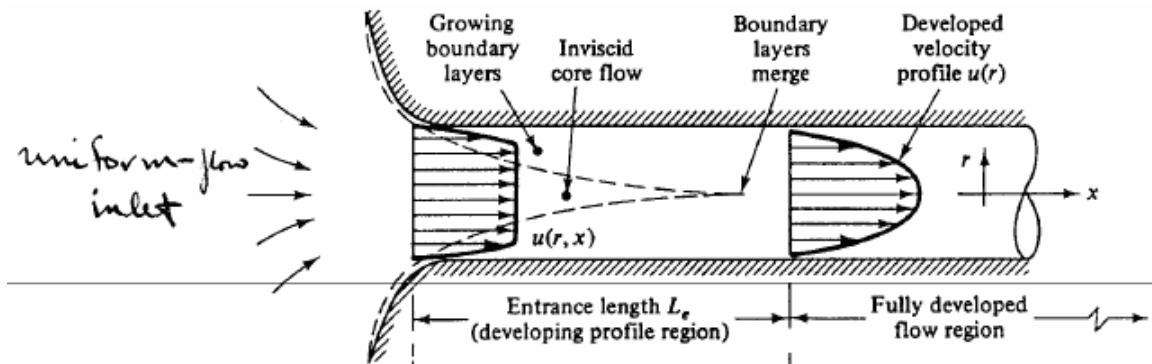
$$\Pi_{i \text{ model}} = \Pi_{i \text{ prototype}}$$

$$i = 1, r = n - \hat{m} \text{ (m)}$$

Similarity is classified as:

1. Geometric Similarity (similar length scale) : $\alpha = L_m / L_p$
2. Kinematic Similarity (similar length and time scale)
3. Dynamic Similarity (similar length, time, and force scales)
(*Re* & *Fr* scaling etc)

Chapter 8: Flow in Pipes



Entrance vs. fully developed flow: $L_e/D = f(\text{Re})$

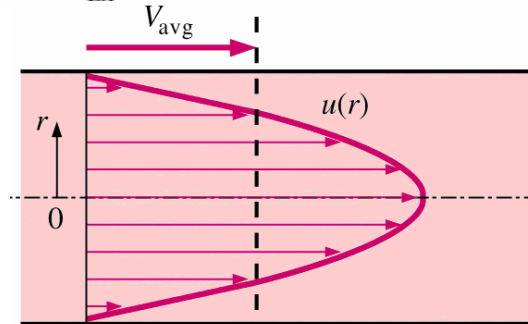
Laminar vs. turbulent flow: $Re_{crit} \approx 2000$

- **Laminar Flow in Pipes**

Velocity profile for laminar pipe flow

$$u(r) = \frac{r^2 - r_0^2}{4\mu} \frac{d}{dx}(p + \gamma z),$$

$$u(r) > 0 \text{ for } \frac{d}{dx}(p + \gamma z) < 0, \text{ i.e., favorable pressure gradient}$$



Laminar flow

Head loss and friction factor for laminar pipe flow (These are exact solutions!!):

$$\frac{p_1}{\gamma} + \frac{V_1^2}{2g} + z_1 = \frac{p_2}{\gamma} + \frac{V_2^2}{2g} + z_2 + h_L$$

$$\Delta h = \left(\frac{p_2}{\gamma} + z_2 \right) - \left(\frac{p_1}{\gamma} + z_1 \right)$$

$$h_L = \frac{p_1 - p_2}{\gamma} + (z_1 - z_2) = -\Delta h$$

$$h_L = \frac{L}{\gamma} \left[-\frac{d}{ds}(p + \gamma z) \right] \quad L = \text{length of pipe} = ds$$

$$= \frac{L}{\gamma} \left[\frac{8\mu \bar{V}}{r_0^2} \right] = -\Delta h \alpha \bar{V} \quad h_L = L \left[-\frac{d}{ds} \left(\frac{p}{\gamma} + z \right) \right]$$

$$= L \left(-\frac{dh}{ds} \right)$$

or $h_f = h_L = \frac{32\mu L \bar{V}}{\gamma D^2}$ $h_f = \text{head loss due to friction}$

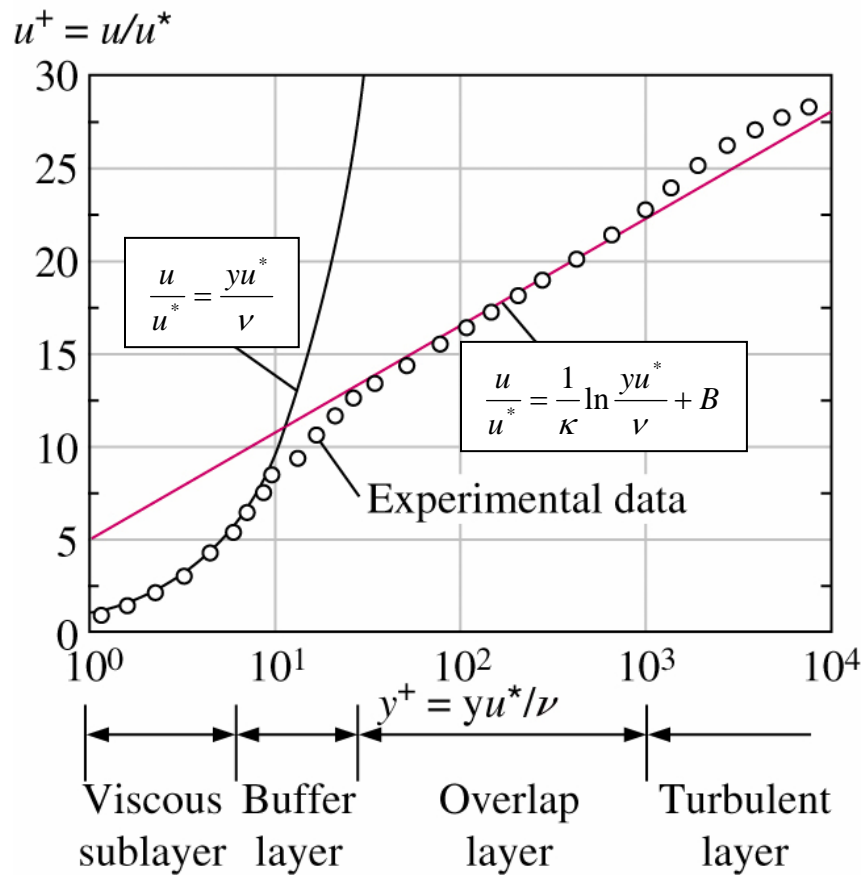
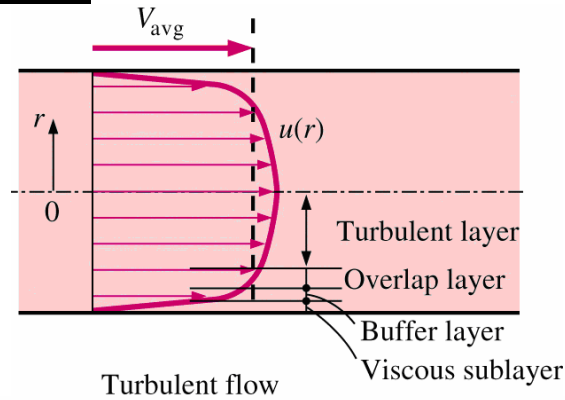
exact solution

$$h_L = \frac{32\mu L V}{\gamma D^2} = f \frac{L V^2}{D 2g}$$

$$\tau_0 = \frac{D \gamma h_L}{4 L} \Rightarrow h_L = \frac{4L \tau_0}{D \gamma} = 4L \frac{1}{8} \rho V^2 f / D \gamma = f \frac{L V^2}{D 2g}$$

$$\therefore f = \frac{64}{Re} \quad Re = \frac{VD}{\nu}$$

- Turbulent Flow in Pipes**



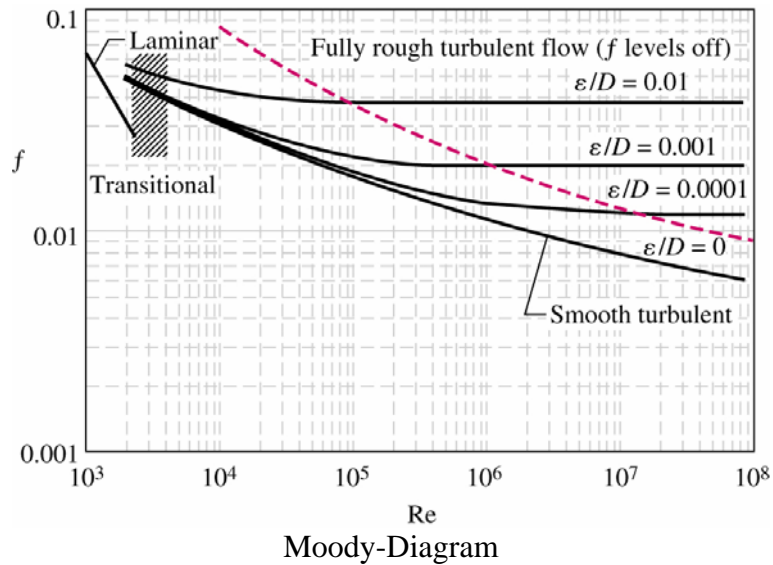
Head losses and friction factor for turbulent pipe flow:

Friction factor:

$$\frac{1}{\sqrt{f}} = 1.14 - 2 \log \left(\frac{k_s/D + 9.35/\text{Re} \sqrt{f}}{\text{Re} \sqrt{f}} \right), \text{ as shown in Moody diagram}$$

k_s/D = roughness parameter

$$h_L = f \frac{L}{D} \frac{V^2}{2g} = h_1 - h_2, \quad h_1 = \frac{P_1}{\gamma} + z_1, \quad h_2 = \frac{P_2}{\gamma} + z_2$$



- **Types of problems for turbulent pipe flow**

1. Determine head loss

$$f = f(\text{Re}, k_s/D)$$

$$h_L = f \frac{L}{D} \frac{V^2}{2g} = h_1 - h_2$$

2. Determine flow rate

$$V = \underbrace{\left[\frac{2gh_L}{LD} \right]}_{\text{known from data}}^{1/2} f^{-1/2}$$

guess $f \Rightarrow V \Rightarrow \text{Re} \Rightarrow f$, repeat to converge

3. Determine pipe size

$$D = \underbrace{\left[\frac{8LQ^2}{\pi^2 gh_L} \right]}_{\text{known from data}}^{1/5} f^{1/5}$$

guess $f \Rightarrow D \Rightarrow \text{Re} \& k_s/D \Rightarrow f$, repeat to converge

- **Minor losses**

$$\frac{p_1}{\gamma} + z_1 + \alpha_1 \frac{V_1^2}{2g} + h_p = \frac{p_2}{\gamma} + z_2 + \alpha_2 \frac{V_2^2}{2g} + h_t + h_L + \sum_i h_m$$

$$h_m = \left(K \frac{V^2}{2g} \right)_i$$

(K : minor loss coefficient \rightarrow Depending on the shape of the pipe inlet/exit/curvature)

